

Structural Steel Pipes

F_y := 260·MPa

E := 200000·MPa

v := 0.3

ϕ_b := 0.9

ϕ_c := 0.85

ϕ_v := 0.85

D := 50·cm

external diameter

t := 6·mm

L := 4·m

K := 1

assume corresponds to real plane of buckling

support := 0

0 for simply supported
1 for clamped

- out of roundness imperfection less than 1%
- if more the checks could become unconservative

$$I := \frac{\pi}{4} \cdot \left[\left(\frac{D}{2} \right)^4 - \left(\frac{D - 2 \cdot t}{2} \right)^4 \right]$$

$$A := \pi \cdot \left[\left(\frac{D}{2} \right)^2 - \left(\frac{D - 2 \cdot t}{2} \right)^2 \right]$$

$$r := \sqrt{\frac{I}{A}}$$

$$S := \frac{I}{\frac{D}{2}}$$

$$Z_p := S \cdot \frac{4}{\pi} \cdot \left(1 + \frac{t}{D} \right)$$

$$G := \frac{E}{2 \cdot (1 + v)}$$

$$Z := 2 \cdot \left(\frac{L}{D} \right)^2 \cdot \left(\frac{D}{t} \right) \cdot \sqrt{1 - v^2}$$

$$C := 0.165$$

Axial Load Strength

$$k_c := \begin{cases} 4 + \frac{3 \cdot Z^2}{\pi^4} & \text{if support} = 1 \\ \frac{12 \cdot Z^2}{\pi^4} & \text{otherwise} \end{cases}$$

critical stress

$$\sigma_{xc} := \left| \begin{array}{l} k_c \cdot \frac{\pi^2 \cdot E}{12 \cdot (1 + \nu^2) \cdot \left(\frac{L}{t}\right)^2} \text{ if } Z < 2.85 \\ \text{otherwise} \\ \left| \begin{array}{l} \frac{2 \cdot C \cdot E}{\frac{D}{t}} \text{ if } 2.85 \leq Z \wedge Z < \frac{1.2 \cdot \left(\frac{D}{t}\right)^2}{C} \\ \frac{\pi^2 \cdot E}{\left(\frac{K \cdot L}{r}\right)^2} \text{ otherwise} \end{array} \right. \end{array} \right|$$

plate buckling

diamond bulge buckling

column buckling

$$F_{y1} := \min \left(\left(\begin{array}{l} F_y \\ \sigma_{xc} \end{array} \right) \right)$$

the interaction between column buckling and local buckling is made by porting Fy (the lesser of Fy or critical stress) to the column buckling equation.

$$\lambda_c := \frac{K \cdot L}{r \cdot \pi} \cdot \sqrt{\frac{F_{y1}}{E}} \quad \lambda_c = 0.26$$

$$\sigma_n := F_{y1} \cdot \left| \begin{array}{l} 0.658^{\left(\lambda_c^2\right)} \text{ if } 0 \leq \lambda_c \leq 1.5 \\ \frac{0.877}{\lambda_c^2} \text{ otherwise} \end{array} \right|$$

$$P_n := \sigma_n \cdot A$$



$$P_n = 239.84 \text{ ton}$$

$$\phi_c \cdot P_n = 203.86 \text{ ton}$$

$$\sigma_n = 0.97 F_y$$

Bending Strength



$$\alpha := \frac{\frac{E}{F_y}}{\frac{D}{t}}$$

$$M_p := Z_p \cdot F_y$$

$$M_y := S \cdot F_y$$

$$M_n := \left| \begin{array}{ll} 0.9 \cdot M_p \cdot \left| \begin{array}{ll} 1 & \text{if } \alpha \leq 14 \\ 0.775 + 0.016 \cdot \alpha & \text{otherwise} \end{array} \right. & \text{if } Z < \frac{1.2 \cdot \left(\frac{D}{t}\right)^2}{C} \\ S \cdot \frac{\pi^2 \cdot E}{\left(\frac{K \cdot L}{r}\right)^2} & \text{otherwise} \end{array} \right.$$



$$M_n = 34.94 \text{ m} \cdot \text{ton}$$

$$\phi_b \cdot M_n = 31.44 \text{ m} \cdot \text{ton}$$

$$M_n = 0.9 M_p$$

$$M_n = 1.16 M_y$$



$$K_s := \begin{cases} \text{if } Z < 100 \\ \begin{cases} 5.35 + 0.213 \cdot Z & \text{if support} = 0 \\ 8.98 + 0.1 \cdot Z & \text{otherwise} \end{cases} \\ \text{otherwise} \\ \begin{cases} 0.85 \cdot Z^{0.75} & \text{if } 100 \leq Z \wedge Z < 19.2 \cdot \left(1 - v^2\right) \cdot \left(\frac{D}{t}\right)^2 \\ \frac{0.406 \cdot Z}{\left(1 - v^2\right)^{0.25} \cdot \left(\frac{D}{t}\right)^{0.5}} & \text{otherwise} \end{cases} \end{cases}$$

strictly it should be <50

$$K_s = 861.15$$

$$\tau_y := \frac{F_y}{\sqrt{3}}$$

$$\tau_{c_t_e} := K_s \cdot \frac{\pi^2 \cdot E}{12 \cdot \left(1 - v^2\right) \cdot \left(\frac{L}{t}\right)^2}$$

$$\tau_{c_t} := \begin{cases} \tau_{c_t_e} & \text{if } \tau_{c_t_e} \leq 0.8 \cdot \tau_y \\ \sqrt{0.8 \cdot \tau_y \cdot \tau_{c_t_e}} & \text{otherwise} \end{cases}$$

$$\tau_{c_v} := 1.25 \cdot \tau_{c_t}$$



Beam-column: Axial load, bending, torsion and shear combined

Factored loads

$P_u := 125 \cdot \text{ton}$

$T_u := 2 \cdot \text{m} \cdot \text{ton}$

$V_u := 10 \cdot \text{ton}$

$M_u := 8 \cdot \text{m} \cdot \text{ton}$

- substitute amplified moment if required
- combine vectorially Mx and My at the point to get Mu

Note that a proper K must be entered in former input above for a proper check.



$$\tau_c := \begin{cases} \tau_{c_v} & \text{if } T_u = 0 \cdot \text{m} \cdot \text{ton} \\ \tau_{c_t} & \text{otherwise} \end{cases}$$

$$\text{Shear_buckling} := \begin{cases} \text{"elastic"} & \text{if } \tau_{c_t} \leq 0.8 \cdot \tau_y \\ \text{"inelastic"} & \text{otherwise} \end{cases}$$



Shear_buckling = "inelastic"

$$\tau_c = 1.37 \tau_y$$

- inelastic shear action is assumed to start at buckling stress = $0.8 \cdot \tau_y$
- the elastic or inelastic critical shear buckling stress, hence



$$\tau_n := \begin{cases} \tau_c & \text{if } \tau_c \leq \tau_y \\ \tau_y & \text{otherwise} \end{cases}$$

this statement establishes that if there's no buckling, the shear yield can be attained



$$\tau_n = 1 \tau_y$$

for most common structural pipes the nominal shear stress will correspond to yield stress



$$\sigma_u := \frac{\frac{P_u}{\phi_c}}{A} + \frac{\frac{M_u}{\phi_b}}{S}$$

$$\tau_u := \frac{\frac{V_u}{A} + \frac{2 \cdot T_u \cdot \frac{D}{2}}{\pi \cdot \left[\left(\frac{D}{2} \right)^4 - \left(\frac{D - 2 \cdot t}{2} \right)^4 \right]}}{\phi_v}$$



$$\frac{\sigma_u}{\sigma_n} + \left(\frac{\tau_u}{\tau_n} \right)^2 = 0.94$$

- must be equal to or less than 1 for the strength to be OK (interaction supported in p.172)
- note that the different strength reduction factors have already been taken into account by enlarging (in the collapsed area) the factored longitudinal and transverse stresses to be checked

- the failure to consider interaction with high shear stresses may be one fundamental cause of failure under earthquake or strong wind forces
- when strong lateral forces are being considered, don't forget to consider shear plus all other actions combined at both the joints and members