

Buckling of symmetric antifunicular T section arches



$L := 50\cdot\text{m}$

span

$f := 20\cdot\text{m}$

sagitta

$f_c := 35\cdot\text{MPa}$

$\gamma_c := 2400\cdot\frac{\text{kgf}}{\text{m}^3}$

$b(y) := \begin{cases} 30\cdot\text{cm} & \text{if } y \leq 58\cdot\text{cm} \\ 222\cdot\text{cm} & \text{otherwise} \end{cases}$

$d := 70\cdot\text{cm}$

total depth

$w := 3\cdot\frac{\text{ton}}{\text{m}}$

$\text{ArchType} := 3$

1 for 3 hinges
2 for 2 hinges
3 for fixed

out of plane fixity is
assumed at supports

Section properties

We will be considering the T beam defined above. In our case the section remains constant along the arch length.

$A_c := \int_{0\cdot\text{m}}^d b(y) \, dy$

$A_c = 4405.35\text{cm}^2$

$\text{own_weight} := A_c\cdot\gamma_c$

$\text{own_weight} = 1.06\frac{\text{ton}}{\text{m}}$

merely information

$y_g := \frac{\int_{0\cdot\text{m}}^d y\cdot b(y) \, dy}{A_c}$

$y_g = 49.6\text{cm}$

$I_x := \int_{0\cdot\text{m}-y_g}^{d-y_g} y^2\cdot b(y) \, dy$

$I_x = 1304896.94\text{cm}^4$

$i_x := \sqrt{\frac{I_x}{A_c}}$

$i_x = 17.21\text{cm}$

Transversally

$$B(y) := \begin{cases} 12\cdot\text{cm} & \text{if } y \leq -15\cdot\text{cm} \\ \text{otherwise} & \\ 12\cdot\text{cm} & \text{if } y \geq 15\cdot\text{cm} \\ 70\cdot\text{cm} & \text{otherwise} \end{cases}$$

$$I_y := \int_{-111\cdot\text{cm}}^{111\cdot\text{cm}} y^2 \cdot B(y) \, dy$$

change limits of integration for your case

we have taken 8 times thickness at each side since we are thinking of arches solidary to equal shape shells, to avoid in such case y buckling controlling, which will not due to impossibility of the arches going out of the plane

$$I_y = 1.09 \times 10^7 \text{ cm}^4$$

$$i_y := \sqrt{\frac{I_y}{A_c}}$$

$i_y = 49.84 \text{ cm}$

Buckling

Source of buckling treatment is *Arcos y Pandeo*, Arangoá, ETSAM 1974
Of course the treatment is conventional and only approximate.

In plane buckling

$$ps := \begin{pmatrix} 0.05 \\ 0.2 \\ 0.3 \\ 0.4 \\ 0.5 \end{pmatrix} \quad \beta1s := \begin{pmatrix} 1.2 \\ 1.16 \\ 1.13 \\ 1.19 \\ 1.25 \end{pmatrix} \quad \beta2s := \begin{pmatrix} 1 \\ 1.06 \\ 1.13 \\ 1.19 \\ 1.25 \end{pmatrix} \quad \beta3s := \begin{pmatrix} 0.7 \\ 0.72 \\ 0.74 \\ 0.75 \\ 0.76 \end{pmatrix}$$

$$vs1 := \text{pspline}(ps, \beta1s) \quad vs2 := \text{pspline}(ps, \beta2s) \quad vs3 := \text{pspline}(ps, \beta3s)$$

$$\beta_1(p) := \text{interp}(vs1, ps, \beta1s, p) \quad \beta_2(p) := \text{interp}(vs2, ps, \beta2s, p) \quad \beta_3(p) := \text{interp}(vs3, ps, \beta3s, p)$$

$$\beta(p) := \left\{ \begin{array}{l} \beta_3(p) \text{ if ArchType} = 3 \\ \text{otherwise} \\ \left\{ \begin{array}{l} \beta_2(p) \text{ if ArchType} = 2 \\ \beta_1(p) \text{ otherwise} \end{array} \right. \end{array} \right.$$

$$\lambda s := \left(\begin{array}{c} 50 \\ 70 \\ 85 \\ 105 \\ 120 \\ 140 \end{array} \right) \qquad \omega s := \left(\begin{array}{c} 1 \\ 1.08 \\ 1.32 \\ 1.72 \\ 2.28 \\ 3 \end{array} \right)$$

$$vs4 := lspline(\lambda s, \omega s) \qquad \omega(\lambda) := \left\{ \begin{array}{l} \text{interp}(vs4, \lambda s, \omega s, \lambda) \text{ if } \text{interp}(vs4, \lambda s, \omega s, \lambda) \geq 1 \\ 1 \text{ otherwise} \end{array} \right.$$

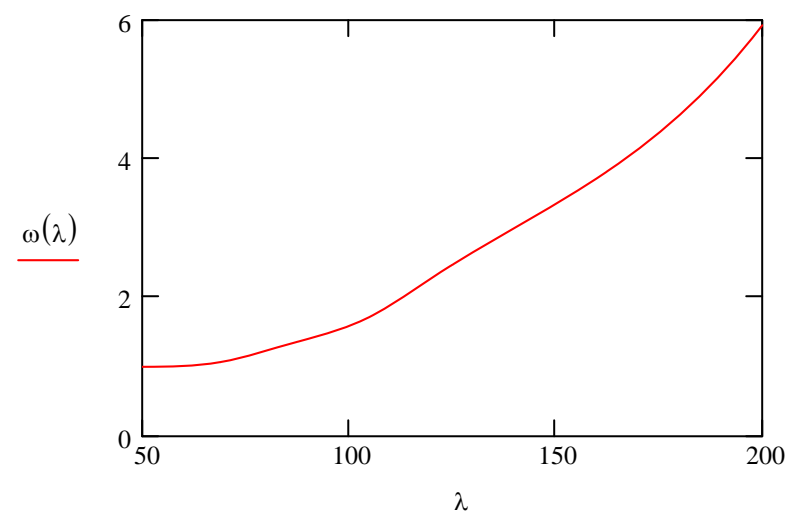
Parabola arch equation (centered) is

$$y(x,f) := -4 \cdot \frac{f}{L^2} \cdot x^2 + f$$

$$\text{slope}(x) := \frac{d}{dx}y(x,f) \qquad \alpha(x) := \text{atan}(\text{slope}(x))$$

$$S_2 := \int\limits_{0 \cdot \text{m}}^{\frac{L}{2}} \sqrt{1 + \left(\frac{d}{dx}y(x,f)\right)^2} \; dx \qquad S_2 = 33.34 \text{m} \qquad \text{semiarch length}$$

$$p_0 := \frac{f}{L} \qquad p_0 = 0.4 \qquad \text{sagitta to span ratio}$$



$$\lambda_x := \beta(p_0) \cdot \frac{S_2}{i_x}$$

$\lambda_x = 145.3$

better don't exceed slenderness 140

$$\omega(\lambda_x) = 3.18$$

Out of plane buckling

Out of Plane Buckling is of not concern **IF** bracing effectively prevents the arches going out of their plane, such when they are part of a shell of same shape. For the other cases...

Type_{arch2} := 1

1 for lx=constant inertia
2 for lx·cosα=constant

$p2s := \begin{pmatrix} 0.05 \\ 0.1 \\ 0.2 \\ 0.3 \\ 0.4 \end{pmatrix}$
 $\beta2s1 := \begin{pmatrix} 0.5 \\ 0.54 \\ 0.65 \\ 0.82 \\ 1.07 \end{pmatrix}$
 $\beta2s2 := \begin{pmatrix} 0.5 \\ 0.52 \\ 0.59 \\ 0.71 \\ 0.86 \end{pmatrix}$

vs5 := pspline(p2s,β2s1)

vs6 := pspline(p2s,β2s2)

$\beta_{2_1}(p) := \text{interp}(vs5, p2s, \beta2s1, p)$
 $\beta_{2_2}(p) := \text{interp}(vs6, p2s, \beta2s2, p)$

$$\beta_2(p) := \left\{ \begin{array}{ll} \beta_{2_1}(p) & \text{if } Type_{arch2} = 1 \\ \beta_{2_2}(p) & \text{otherwise} \end{array} \right.$$

$\lambda_y := \beta_2\left(\frac{f}{L}\right) \cdot \frac{L}{i_y}$

$\lambda_y = 107.35$

better don't exceed 140

$\omega(\lambda_y) = 1.8$

$$\omega := \max\left(\left(\omega(\lambda_x)\right), \left(\omega(\lambda_y)\right)\right)$$

$\omega = 3.18$

curvature

$$\kappa(x) := \frac{\frac{d^2}{dx^2}y(x,f)}{\left[1 + \left(\frac{d}{dx}y(x,f)\right)^2\right]^{\frac{3}{2}}}$$

$$R(x) := \frac{1}{\kappa(x)}$$

radius of curvature

These two equations make nothing but to keep information available

$weight(X) := 2 \cdot X \cdot w$

vertical action of symmetric central cut, origin at center

$$N_\phi(x) := \frac{\frac{weight(x)}{\cos(atan(slope(x)))}}{2}$$

tangential force

At ends

$x := \frac{L}{2}$
 $\sigma_{\phi}(x) := \frac{N_{\phi}(x)}{A_c}$

$N_{\phi}(x) = 141.51 \text{ ton}$
 $\sigma_{\phi}(x) = 3.15 \text{ MPa}$
 $\alpha(x) = -57.99 \text{ deg}$

At 1/4th span

$x := \frac{L}{4}$
 $\sigma_{\phi}(x) := \frac{N_{\phi}(x)}{A_c}$

$N_{\phi}(x) = 48.02 \text{ ton}$
 $\sigma_{\phi}(x) = 1.07 \text{ MPa}$
 $\alpha(x) = -38.66 \text{ deg}$

Axial Stress that controls

$$\sigma_{\phi_max} := \max \left(\left(\sigma_{\phi} \left(\frac{L}{2} \right) \right) \left(\omega \cdot \sigma_{\phi} \left(\frac{L}{4} \right) \right) \right)$$

$\sigma_{\phi_max} = 3.4 \text{ MPa}$

$\frac{\sigma_{\phi_max}}{f_c} = 9.72 \%$

OK, buckling seems of not concern

I wouldn't let go it over 40% in non seismic zones and over 25% in seismic zones, without further consideration of concomitant bending

- The stress at 1/4 point would be combined with the strength of materials derived stresses from bending in x direction (and y, if significant).
- At supports a common strength of materials check (without amplification due to buckling) seems to be enough.

Reactions (if hinged)

$$\text{Reaction} := N_{\phi}\left(\frac{L}{2}\right) \quad R_v := -N_{\phi}\left(\frac{L}{2}\right) \cdot \sin(\alpha(x)) \quad R_h := N_{\phi}\left(\frac{L}{2}\right) \cdot \cos(\alpha(x))$$

$$\text{Reaction} = 141.51 \text{ ton}$$

$$R_v = 88.4 \text{ ton}$$

$$R_h = 110.5 \text{ ton}$$

Chart parabolic arch

$$\text{Parts} := 100 \quad j := 1 \,..\, \text{Parts} + 1 \quad x_j := -\frac{L}{2} + (j - 1) \cdot \frac{L}{\text{Parts}} \quad y_j := y(x_j, f)$$

