

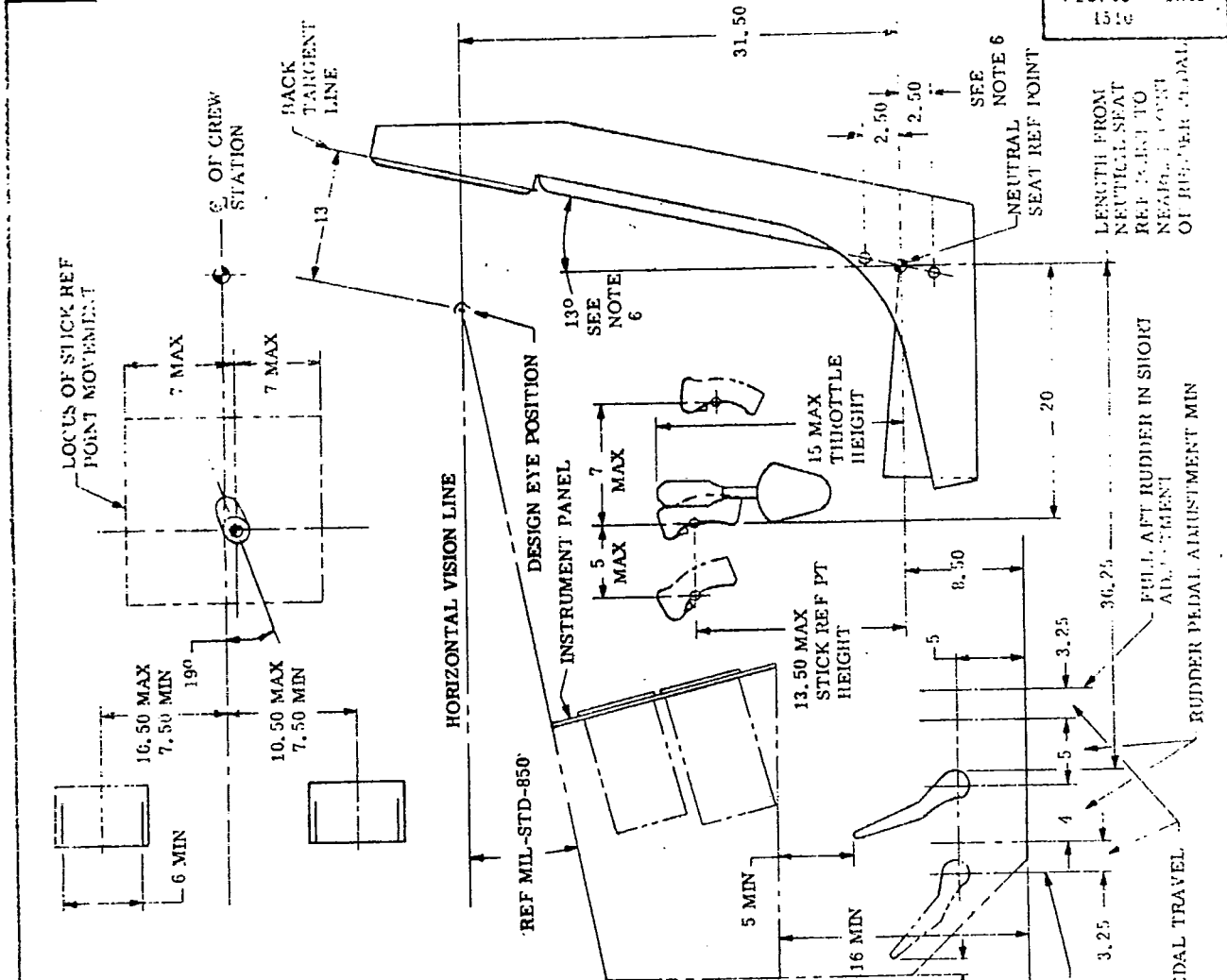
USER SYMBOLS:

REVERSE SYMBOLS:

Information is current as of the date of this document. Any future coordination of changes to this document, draft circulation will be based on the information in the current DOCSIS.

1. ALL MEASUREMENTS ARE BASED UPON THE SEAT REFERENCE POINT AT THE CENTERLINE OF THE SEAT IN THE NEUTRAL POSITION.
2. REFERENCE SPECIFICATION MIL-STD-8534 FOR BRAKE PEDAL ANGLES AND DIMENSIONS.
3. THE STICK AND THROTTLE REFERENCE POINT IS DEFINED AS THE POINT AT WHICH THE PILOT'S SECOND FINGER IS IN CONTACT WITH THE FORWARD FACE OF THE CONTROL.
4. RUDDER PEDAL ADJUSTMENT SHALL BE IN INCREMENTS OF 1 INCH OR LESS.
5. BECAUSE OF CATAPULT GRIP REQUIREMENT, MAX. FORWARD THROTTLE MOVEMENT FOR AIRCRAFT EQUIPPED FOR CATAPULTING SHALL BE 20 INCHES FROM NEUTRAL SEAT REFERENCE POINT, MAX.
6. FORWARD THROTTLE MOVEMENT FOR NONCATAPULT AIRCRAFT SHALL CORRESPOND TO THAT OF STICK. FOR SEAT BACK ANGLE AND SEAT ADJUSTMENT REQUIREMENTS FOR EJECTION SEATS IN U.S. NAVY AIRCRAFT, MIL-S-18471 AND NAVAIR 65A136H1 APPLY.
7. DIMENSIONS IN INCHES.

THIS IS A DESIGN STANDARD, NOT USED AS A PART NUMBER.



P.A. NAVY - AS Other Cust ARMY - AV AF - 11 DEPARTMENT SPECIFICATION (20)	TITLE DIMENSIONS, BASIC, COCKPIT, STICK CONTROLLED, FIXED WING AIRCRAFT SUPERSEDES:	MILITARY STANDARD MS33574 SHEET 1 of 1
--	--	--

ENTIRE STD REVISED AND RE-DRAWN
 APPROVED 24 JULY 56 REVISED (A) 5 JUN 57 2 JUN 59