

BRIDGE CAPACITY RATING

NEW YORK

DEPARTMENT OF TRANSPORTATION

LONG ISLAND EXPRESSWAY

OVER

CEMETARY ROAD UNDERPASS

September, 1998

**RUST**

**Rust Environment  
& Infrastructure**

**LONG ISLAND EXPRESSWAY  
OVER  
CEMETARY ROAD UNDERPASS**

September 16, 1998

**COMPUTATION INDEX**

	PAGE NO.
BRIDGE CAPACITY RATING SUMMARY SHEET	
RIGID FRAME DIMENSIONS . . . . .	1
RIGID FRAME INPUT COORDINATES . . . . .	2
RIGID FRAME PROPERTIES . . . . .	3 - 6
RIGID FRAME LOADS . . . . .	7 - 12
DEAD LOAD STRESSES . . . . .	13 - 16
LIVE LOAD FORCES . . . . .	17 - 20
BEAM PROPERTIES . . . . .	21 - 22
RATING COMPUTATIONS . . . . .	23 - 24
DEAD AND LIVE LOAD STRESSES. . . . .	25 - 36
STAAD DEAD LOAD ANALYSIS. . . . .	37 - 50
STAAD LIVE LOAD ANALYSIS . . . . .	51 - 62
LIVE LOAD INFLUENCE LINES. . . . .	63 - 67

# BRIDGE CAPACITY RATING

September 16, 1998

## Summary

### Reinforced Concrete Rigid Frame Rating Analysis - HS-20 Loading

<u>Location</u>	<u>Inventory Rating (Tons)</u>	<u>Equivalent HS Loading (Tons)</u>	<u>Inventory Rating (Tons)</u>	<u>Equivalent HS Loading (Tons)</u>
Abutment Face	38.8	HS 21.5	94.7	HS 52.6
1 / 8 th Point	87.0	HS 48.3	142.8	HS 79.3
1 / 4 Point	83.1	HS 46.2	120.0	HS 66.7
3 / 8 th Point	52.8	HS 29.3	84.6	HS 47.0
Crown	35.3	HS 19.6	69.6	HS 38.7

CALCULATION SHEET

PAGE 1 OF     

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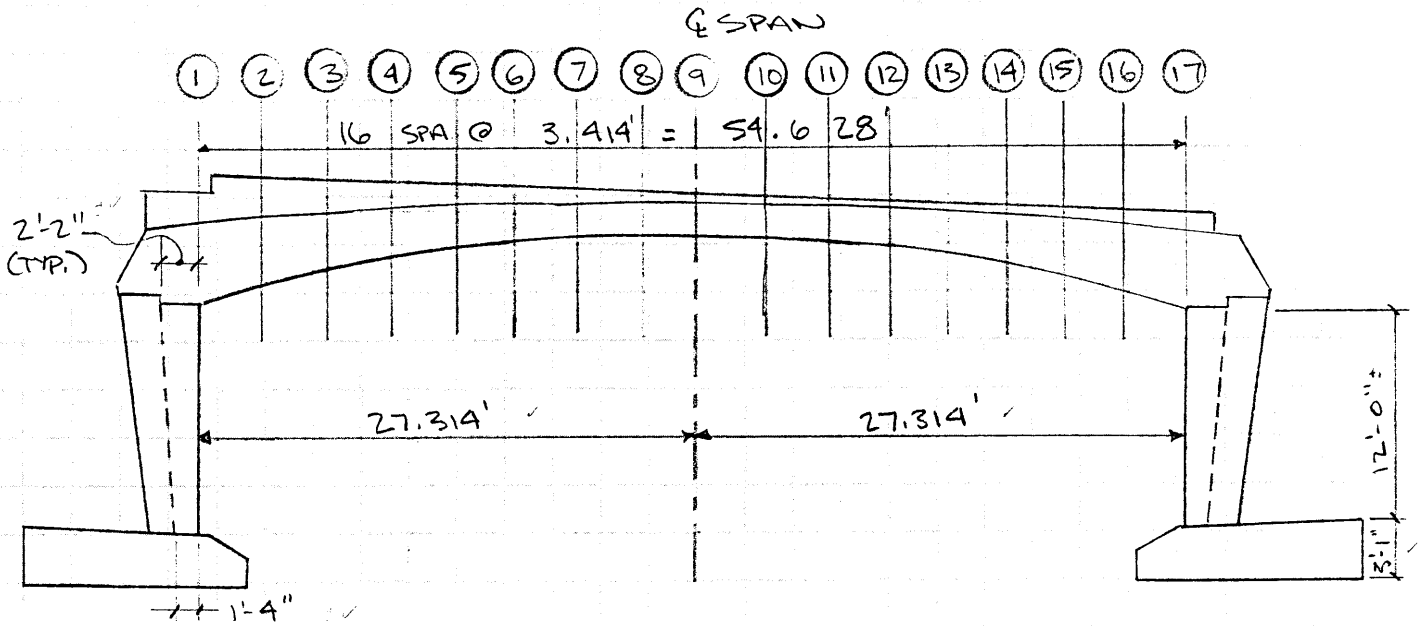
CLIENT N.Y. DOT SUBJECT CEMETERY

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PAGE 2 OF     

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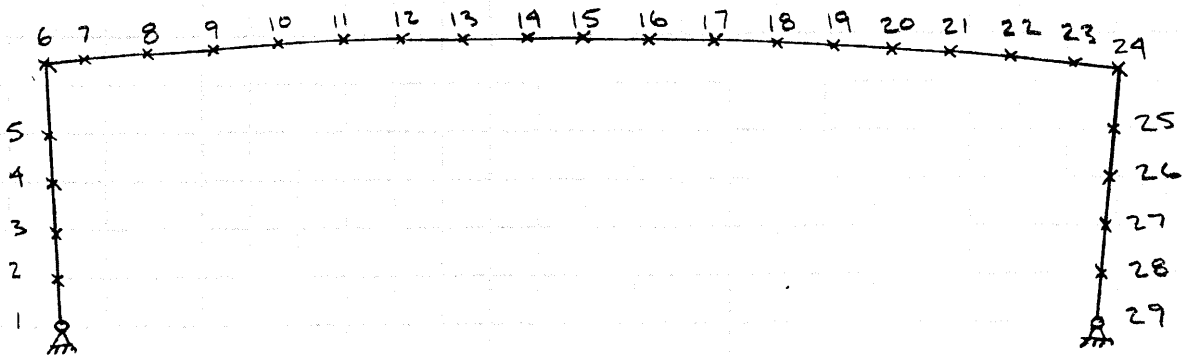
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COORDINATES FOR STAAD INPUT



JOINT #	X	Y	Z	JOINT #	X	Y	Z
1	0.0	0.0	0.0 ✓	16	32.06	17.09	0.0
2	-0.23	3.33	0.0	17	35.48	16.95	0.0
3	-0.43	6.33	0.0	18	38.89	16.88	0.0
4	-0.63	9.33	0.0	19	42.30	16.97	0.0
5	-0.83	12.33	0.0	20	45.72	16.20	0.0
6	-0.95	19.06	0.0 ✓	21	49.13	15.75	0.0
7	1.33	19.58	0.0	22	52.55	15.21	0.0
8	4.75	15.21	0.0	23	55.96	14.58	0.0
9	8.16	15.75	0.0	24	58.25	14.06	0.0 ✓
10	11.58	16.20	0.0	25	58.13	12.33	0.0
11	14.99	16.47	0.0	26	57.93	9.33	0.0
12	18.40	16.88	0.0	27	57.72	6.33	0.0
13	21.82	16.95	0.0	28	57.52	3.33	0.0
14	25.23	17.09	0.0	29	57.29	0	0.0 ✓
15	28.65	17.16	0.0				

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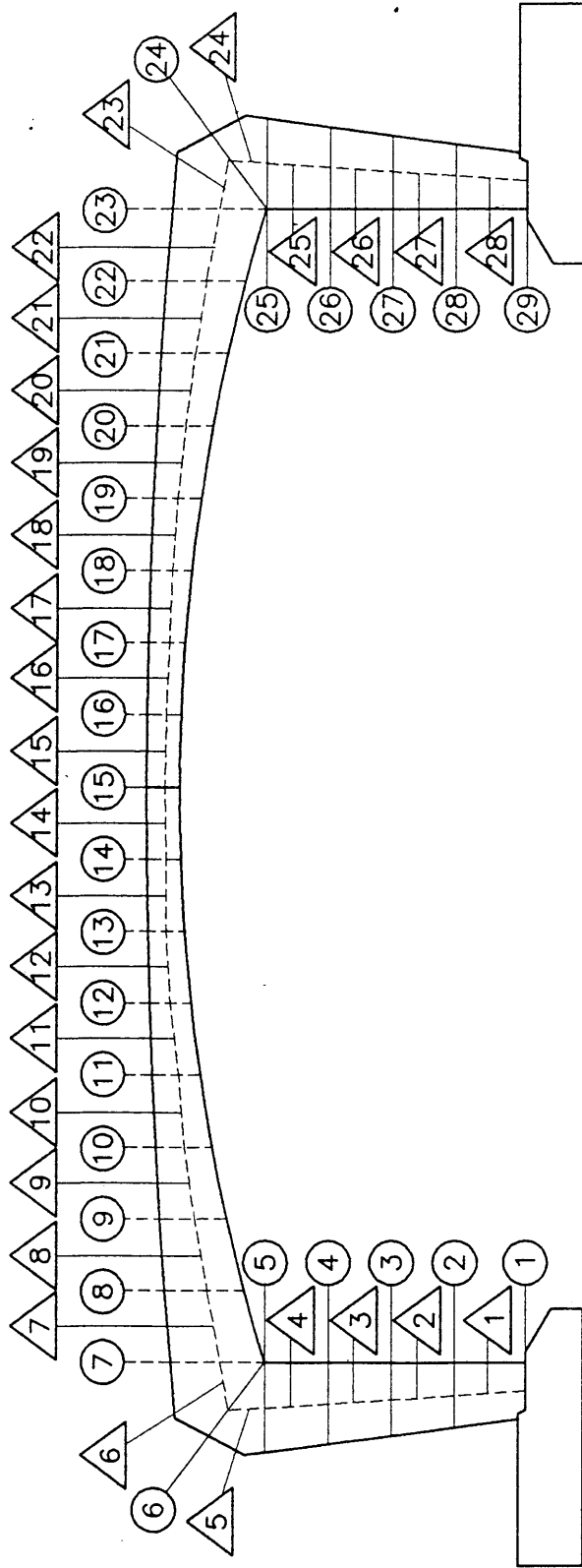
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SECTION PROPERTIES FOR 1'-0" STRIP OF BRIDGE.

JOINT #	AREA (FT <sup>2</sup> )	WT / FT (K/FT) @ .150 K/FT <sup>2</sup>
1	2.62	0.393
2	3.07	0.461
3	3.47	0.522
4	3.88	0.582
5	4.28	0.642
6	4.42	0.663
7	4.50	0.675
8	3.80	0.570
9	3.20	0.480
10	2.70	0.405
11	2.30	0.345
12	1.98	0.297
13	1.76	0.264
14	1.63	0.245
15	1.58	0.237
16	1.63	0.245
17	1.76	0.264
18	1.98	0.297
19	2.30	0.345
20	2.70	0.405
21	3.20	0.480
22	3.80	0.570
23	4.50	0.675
24	4.42	0.663
25	4.28	0.642
26	3.88	0.582
27	3.47	0.522
28	3.07	0.461
29	2.62	0.393



△ 7 MEMBER NUMBER

⊙ 5 JOINT NUMBER

CEMETERY ROAD UNDERPASS

Moments of Inertia

*Pages*

Joint #	Member #	b(ft.)	h(ft.)	lx(in <sup>4</sup> )	Average lx(in <sup>4</sup> )	J(in <sup>4</sup> )	Average J(in <sup>4</sup> )
1	1	1.00	2.62	31078	40538	124310	162153
2	2	1.00	3.07	49999	61099	199995	244396
3	3	1.00	3.47	72199	86567	288797	346267
4	4	1.00	3.88	100934	118207	403737	472829
5	5	1.00	4.28	135480	142347	541920	569389
6	6	1.00	4.42	149214	153339	596857	613357
7	7	1.00	4.50	157464	126141	629856	504566
8	8	1.00	3.80	94819	75721	379275	302884
9	9	1.00	3.20	56623	45318	226492	181271
10	10	1.00	2.70	34012	27518	136049	110074
11	11	1.00	2.30	21025	17219	84098	68876
12	12	1.00	1.98	13413	11417	53654	45668
13	13	1.00	1.76	9421	8452	37683	33808
14	14	1.00	1.63	7484	7150	29934	28599
15	15	1.00	1.58	6816	7150	27263	28599
16	16	1.00	1.63	7484	8452	29934	33808
17	17	1.00	1.76	9421	11417	37683	45668
18	18	1.00	1.98	13413	17219	53654	68876
19	19	1.00	2.30	21025	27518	84098	110074
20	20	1.00	2.70	34012	45318	136049	181271
21	21	1.00	3.20	56623	75721	226492	302884
22	22	1.00	3.80	94819	126141	379275	504566
23	23	1.00	4.50	157464	153339	629856	613357
24	24	1.00	4.42	149214	142347	596857	569389
25	25	1.00	4.28	135480	118207	541920	472829
26	26	1.00	3.88	100934	86567	403737	346267
27	27	1.00	3.47	72199	61099	288797	244396
28	28	1.00	3.07	49999	40538	199995	162153
29	29	1.00	2.62	31078		124310	



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19-Aug-98

*Page 6*

Joint #	Member #	AREA(ft.^2)	AREA(IN.^2)	Average AREA	WT.(lb/ft.)	Average WT.
1	1	2.62	377	410	0.393	0.43
2	2	3.07	442	471	0.461	0.49
3	3	3.47	500	529	0.521	0.55
4	4	3.88	559	588	0.582	0.61
5	5	4.28	616	626	0.642	0.65
6	6	4.42	636	642	0.663	0.67
7	7	4.50	648	598	0.675	0.62
8	8	3.80	547	504	0.570	0.53
9	9	3.20	461	425	0.480	0.44
10	10	2.70	389	360	0.405	0.38
11	11	2.30	331	308	0.345	0.32
12	12	1.98	285	269	0.297	0.28
13	13	1.76	253	244	0.264	0.25
14	14	1.63	235	231	0.245	0.24
15	15	1.58	228	231	0.237	0.24
16	16	1.63	235	244	0.245	0.25
17	17	1.76	253	269	0.264	0.28
18	18	1.98	285	308	0.297	0.32
19	19	2.30	331	360	0.345	0.38
20	20	2.70	389	425	0.405	0.44
21	21	3.20	461	504	0.480	0.53
22	22	3.80	547	598	0.570	0.62
23	23	4.50	648	642	0.675	0.67
24	24	4.42	636	626	0.663	0.65
25	25	4.28	616	588	0.642	0.61
26	26	3.88	559	529	0.582	0.55
27	27	3.47	500	471	0.521	0.49
28	28	3.07	442	410	0.461	0.43
29	29	2.62	377		0.393	

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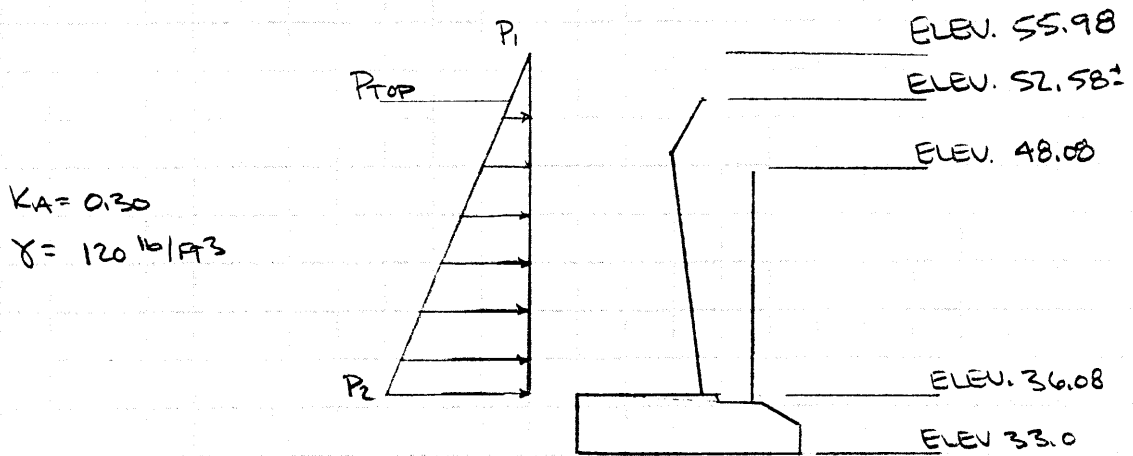
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SOIL PRESSURE ON FRAME



$P_1 = 0$

$P_2 = K_A \gamma H = 0.30 \times 120 \text{ lb/ft}^3 \times 19.9' = 716.4 \text{ lb/ft}^2$

USE  $1/2 P_2$  AS PER AASHTO 3.20.2

$P_2 = 1/2 (716 \text{ lb/ft}^2) = 358.2 \text{ lb/ft}^2$

APPLY TO BOTH SIDES

$P_{\text{TOP OF FRAME}} = \frac{358.2}{19.9} \times (19.9 - 16.5) = 61.2 \text{ lb/ft}^2$

CLIENT N.Y. DOT SUBJECT CEMETERY ROAD

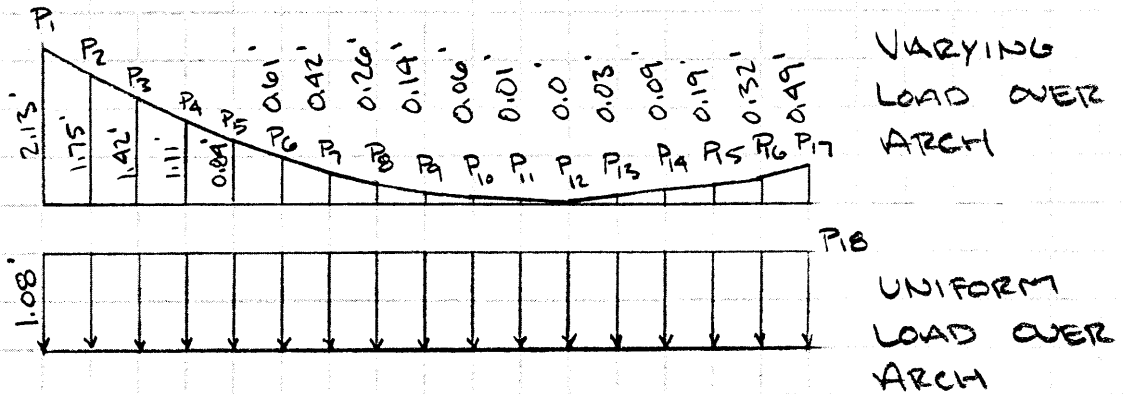
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DEAD LOADS - FILL OVER ARCH + PAVEMENT



- $P_1 = 150 \text{ lb/ft}^3 \times 2.13' \times 1.0' = 319.5 \text{ lbs/ft}$
- $P_2 = 150 \text{ lb/ft}^3 \times 1.75' \times 1.0' = 262.5 \text{ lbs/ft}$
- $P_3 = 150 \text{ lb/ft}^3 \times 1.42' \times 1.0' = 213.0 \text{ lbs/ft}$
- $P_4 = 150 \text{ lb/ft}^3 \times 1.11' \times 1.0' = 166.5 \text{ lbs/ft}$
- $P_5 = 150 \text{ lb/ft}^3 \times 0.84' \times 1.0' = 126.0 \text{ lbs/ft}$
- $P_6 = 150 \text{ lb/ft}^3 \times 0.61' \times 1.0' = 91.5 \text{ lbs/ft}$
- $P_7 = 150 \text{ lb/ft}^3 \times 0.42' \times 1.0' = 63.0 \text{ lbs/ft}$
- $P_8 = 150 \text{ lb/ft}^3 \times 0.26' \times 1.0' = 39.0 \text{ lbs/ft}$
- $P_9 = 150 \text{ lb/ft}^3 \times 0.14' \times 1.0' = 21.0 \text{ lbs/ft}$
- $P_{10} = 150 \text{ lb/ft}^3 \times 0.06' \times 1.0' = 9.0 \text{ lbs/ft}$
- $P_{11} = 150 \text{ lb/ft}^3 \times 0.01' \times 1.0' = 1.5 \text{ lbs/ft}$
- $P_{12} = 150 \text{ lb/ft}^3 \times 0.0' \times 1.0' = 0$
- $P_{13} = 150 \text{ lb/ft}^3 \times 0.03' \times 1.0' = 4.5 \text{ lbs/ft}$
- $P_{14} = 150 \text{ lb/ft}^3 \times 0.09' \times 1.0' = 13.5 \text{ lbs/ft}$
- $P_{15} = 150 \text{ lb/ft}^3 \times 0.19' \times 1.0' = 28.5 \text{ lbs/ft}$
- $P_{16} = 150 \text{ lb/ft}^3 \times 0.32' \times 1.0' = 48.0 \text{ lbs/ft}$
- $P_{17} = 150 \text{ lb/ft}^3 \times 0.49' \times 1.0' = 73.5 \text{ lbs/ft}$
- $P_{18} = 150 \text{ lb/ft}^3 \times 1.08' \times 1.0' = 162.0 \text{ lbs/ft}$

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PAGE 9 OF     

PROJECT NO. 19116

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DEAD LOADS CONT.

BARRIERS & SPLASH GUARD

$$\text{MEDIAN} = (1' \times 3.5' + \frac{1}{2} \times 1.33' \times 3.5') \times 150 \text{ lb/ft}^3 = 795 \text{ #/ft}$$

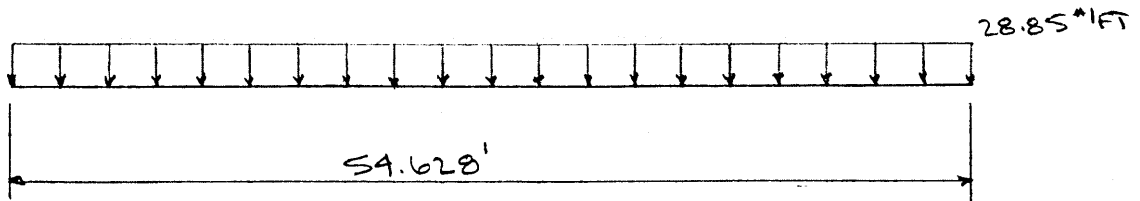
$$\text{PARAPET FACING} = (1.10' \times 3.5' + \frac{1}{2} \times .67' \times 3.5') \times 2 \times 150 \text{ lb/ft}^3 = 1506 \text{ #/ft}$$

SPLASH GUARD = SAY = 19 #/ft

TOTAL = 2320 #/ft

\* DISTRIBUTE OVER ENTIRE WIDTH \*

$$\frac{2320 \text{ #/ft}}{80.42'} = 28.85 \text{ #/ft} / \text{width}$$



CLIENT N.Y. DOT SUBJECT CEMETERY ROAD

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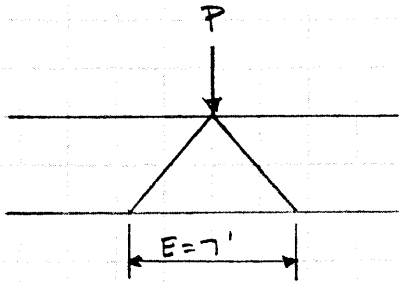
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LIVE LOAD ANALYSIS

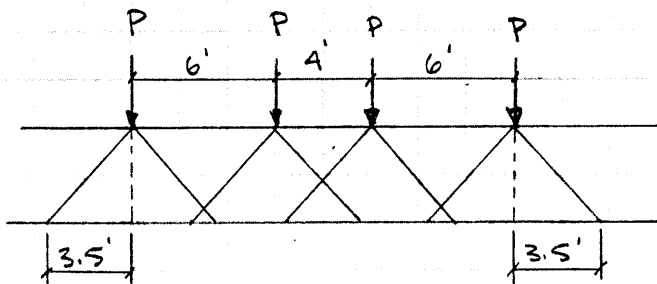
DISTRIBUTION FACTOR - AASHTO 3.24.3.2

$$E = (4 + 0.06S) = 4 + 0.06(54.628') = 7.28' \therefore$$



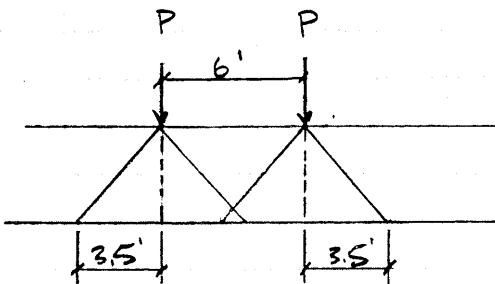
USE 7.0'

TWO TRUCKS



$$D.F. = \frac{4P}{23'} = 0.174P$$

ONE TRUCK



$$D.F. = \frac{2P}{13} = 0.154P$$

TWO TRUCKS CONTROL: USE 0.174P

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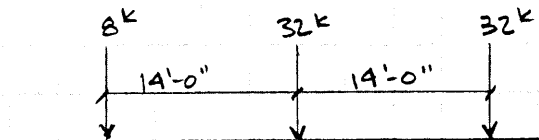
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LIVE LOAD ANALYSIS CONT.

HS20-44 TRUCK



IMPACT  
FOR JOINTS I  
7-14 10%  
15-23 20%

DETERMINE MAXIMUM MOMENT AT FACE OF FRAME, JOINT 7. USING THE INFLUENCE LINE DEVELOPED FOR JOINT 7.

TRY TRUCK @ JOINT 15

$32k \times 8.20$	$=$	$262.4^{1-k}$	I	
$32k \times 5.54$	$=$	$177.3^{1-k}$	1.2	$314.9^{1-k}$
$8k \times 4.90$	$=$	$39.2^{1-k}$	1.2	$212.8^{1-k}$
		<u><math>478.7^{1-k}</math></u>	1.1	<u><math>43.1^{1-k}</math></u>
				$570.8^{1-k}$

TRY TRUCK @ JOINT 14

$32k \times 7.91$	$=$	$253.1^{1-k}$	I	
$32k \times 6.60$	$=$	$211.2^{1-k}$	1.1	$278.4^{1-k}$
$8k \times 3.59$	$=$	$28.3^{1-k}$	1.2	$253.4^{1-k}$
		<u><math>492.6^{1-k}</math></u>	1.1	<u><math>31.1^{1-k}</math></u>
				$562.9^{1-k}$

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PAGE 12 OF     

PROJECT NO. 19116

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LIVE LOAD ANALYSIS CONT.

LL @ FACE

TRY TRUCK @ JOINT 13

			I	
$32^k \times 7.23$	=	$231.4^k$	1.1	$254.5^k$
$32^k \times 7.46$	=	$238.7^k$	1.2	$286.4^k$
$8^k \times 2.10$	=	$16.8^k$	1.1	$18.5^k$
		$486.9^k$		$559.4^k$

TRY TRUCK @ JOINT 16

			I	
$32^k \times 8.07$	=	$258.2^k$	1.2	$309.8^k$
$32^k \times 6.11$	=	$195.5^k$	1.1	$215.1^k$
$8^k \times 4.34$	=	$34.7^k$	1.2	$41.6^k$
		$488.4^k$		$566.5^k$

∴ TRUCK @ JOINT 15 CONTROLS  $570.8^k$

Roll load wheel load

$$M_u = 570.8^k \times 0.174 \times \frac{1}{2} = 49.66^k$$

AXIAL LIVE LOAD FORCE

$$P = (32^k \times 0.70)^{1.20} + (32^k \times 0.45)^{1.20} + (8^k \times 0.57)^{1.10} = 49.2^k$$

$$P_u = 49.2^k \times 0.174 \times \frac{1}{2} = 4.28^k$$

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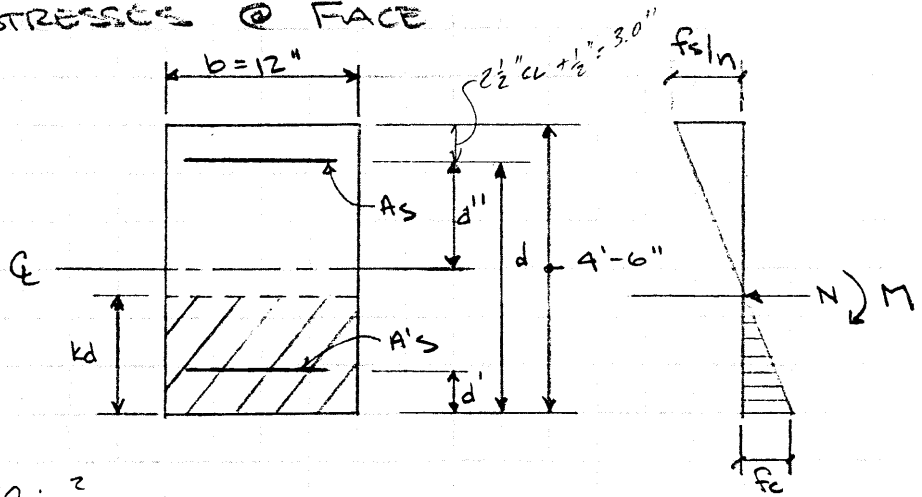
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DEAD LOAD STRESSES @ FACE

- $f_c = 2500 \text{ psi}$
- $n = 12$
- $b = 12 \text{ in}$
- $d = 51.0 \text{ ''}$
- $d' = 2.875 \text{ ''}$
- $d'' = 24.0 \text{ ''}$
- $M_{DL} = 152.65 \text{ FT-K}$
- $P_{DL} = 17.13 \text{ K}$
- $A_s = 1'' \square @ 5'' \text{ O.C} = 2.40 \text{ in}^2$
- $A'_s = 3/4'' \phi @ 10'' \text{ O.C} = 0.53 \text{ in}^2$



$$e = \frac{12 \text{ M}}{N} + d'' = \frac{12 \times 152.65}{17.13} + 24 = 130.9 \text{ in}$$

$$e/d = 130.9 / 51 = 2.57 \quad \text{Assume } j = 0.87$$

From Table 10a Reinforced Concrete Design Handbook  $i \approx +.48$  or

$$i = \frac{1}{1 - \frac{j d}{e}} = \frac{1}{1 - \frac{0.87 \times 51}{130.9}} = 1.51 \quad \checkmark$$

$$m = \frac{n A_s i}{b d} + \frac{(2n-1) A'_s}{b d} = \frac{12 \times 2.40 \times 1.51}{12 \times 51} + \frac{(23) \times 0.53}{12 \times 51}$$

$$m = 0.071 + 0.020 = 0.091$$



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DEAD LOAD STRESSES @ ABUTMENT FACE

$$e = \frac{n A_s i}{b d} + \frac{(2n-1) A_s'}{b d} \times \frac{d'}{d} = 0.071 + 0.020 \left( \frac{2.875}{51} \right) = 0.072$$

$$k = \sqrt{m^2 + 2e} - m = \sqrt{(0.091)^2 + 2(0.072)} - 0.091 = 0.299$$

$$\frac{1}{k} \times \frac{(2n-1) A_s'}{b d} = \frac{1}{0.299} (0.020) = 0.067$$

$$\frac{1}{k} \times \frac{d'}{d} = \frac{1}{0.299} \left( \frac{2.875}{51} \right) = 0.189$$

From Table 12  $z \approx 0.32$

$$\text{OR } z = \frac{\frac{1}{6} + \frac{(2n-1) A_s'}{k b d} \times \frac{d'}{k d} \times \left(1 - \frac{d'}{k d}\right)}{\frac{1}{2} \frac{(2n-1) A_s'}{k b d} \times \left(1 - \frac{d'}{k d}\right)} = \frac{\frac{1}{6} + \frac{0.020}{0.299} \times 0.189 (1 - 0.189)}{\frac{1}{2} + \frac{0.020}{0.299} \times (1 - 0.189)}$$

$$z = 0.319$$

$$j = 1 - z k = 1 - 0.319 (0.299) = 0.905$$

$$\text{Revised } i = \frac{1}{1 - \frac{j d}{e}} = \frac{1}{1 - \frac{0.905 \times 51}{130.9}} = 1.545$$

$$f_s = \frac{17.4^4 \times (1000)}{j A_s i} \times \frac{1}{d} = \frac{17,130}{0.905 \times 2.40 \times 1.545} \times 2.57 = 13,119 \text{ PSI}$$

$$f_c = \frac{f_s}{n} \times \frac{k}{1-k} = \frac{13,119}{12} \times \frac{0.299}{1-0.299} = 466 \text{ PSI}$$

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PAGE 15 OF     

PROJECT NO. 19116

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STRESSES @ ABUTMENT FACE

BY INSPECTION STEEL STRESSES CONTROL

AVAILABLE LIVE LOAD CAPACITY @ ABUTMENT FACE

$$f_s = 18,000 - 13,119 = 4,881 \text{ psi INVENTORY}$$

$$f_s = 25,000 - 13,119 = 11,881 \text{ psi OPERATING}$$

LIVE LOAD STRESSES

$$M_{LL+I} = 49.66 \text{ H-K} \quad P = 4.28^{\text{K}}$$

$$e = 49.66 \times 12 / 4.28 + 24'' = 163.2$$

Assume  $j = 0.907$

$$i = \frac{1}{1 - \frac{0.907 \times 51}{163.2}} = 1.396$$

$$m = \frac{n A_s i}{b d} + \frac{(2n-1) A_s'}{b d} = \frac{12 \times 2.40 \times 1.396}{12 \times 51} + \frac{(23) 0.53}{12 \times 51}$$

$$m = 0.0657 + 0.0200 = 0.0857$$

$$g = 0.0657 + 0.0200 \left( \frac{2.875}{51} \right) = 0.0668$$

$$k = \sqrt{(0.0857)^2 + 2(0.0668)} - 0.0857 = 0.290$$

$$\frac{d'}{h d} = \frac{2.875}{.290 \times 51}$$

$$\frac{d'}{h d} = 0.194$$

## CALCULATION SHEET

PAGE 16 OF     PROJECT NO. 19116CLIENT NY DOT SUBJECT CONCRETE ROADPrepared By JM Date 8/98PROJECT LIC UNDERPASSReviewed By W.V. Date     Approved By      Date     

$$z = \frac{\frac{1}{6} + \frac{0.0200}{0.290} \times 0.194 (1-0.194)}{\frac{1}{2} + \frac{0.0200}{0.290} \times (1-0.194)} = 0.319$$

$$j = 1 - zk = 1 - 0.319(0.290) = 0.907$$

$$f_s = \frac{4280}{0.907 \times 2.40 \times 1.396} \times \frac{163,2}{51} = 4507 \text{ psi}$$

$$f_s \text{ Road Load} + \text{Line Load} = 13,119 + 4507 = 17626$$

$$17626 < 18,000 \therefore \text{OK} \checkmark$$

CALCULATION SHEET

PAGE 17 OF     

PROJECT NO. 19116

CLIENT NY DOT

SUBJECT Concrete Floor

Prepared By Jm Date 9/88

PROJECT L1E

UNDERPASS

Reviewed By WV Date 9/93

Approved By      Date     

LIVE LOAD @ 1/8" Panel

$$M_u = (16' \times 1.20 (4.78 + 5.02) + 4' \times 1.20 (1.18)) \times 0.174 = 33.73 \text{ ft-k}$$

$$P_u = (16' \times 1.20 (0.66 + 0.52) + 4' \times 1.20 (0.12)) \times 0.174 = 4.04$$

DEAD LOAD MOMENT = ~~25~~ 61.35 ft-k  
AXIAL LOAD = 13.73 k

CALCULATION SHEET

PAGE 18 OF     

PROJECT NO. 19116

CLIENT NY DOT

SUBJECT COMETARY ROAD

Prepared By JM Date 9/98

PROJECT LIE

UNDERPASS

Reviewed By JM Date 9/98

Approved By      Date     

LIVE LOAD @  $\frac{1}{4}$  Point

$$M \text{ (POSITIVE)} = 16^4 (4.28) \times 1.20 \times 0.174 = +14.30 \text{ FT-K}$$

$$P = 16 (0.38) \times 1.20 \times 0.174 = 1.27 \text{ K}$$

$$M \text{ (NEGATIVE)} = 16^4 (\cancel{3.29} 3.17 + 2.27) \times 1.2 \times 0.174 = -18.17 \text{ FT-K}$$

$$16 (0.63 + 0.33) \times 1.2 \times 0.174 = 3.21 \text{ K}$$

DEAD LOAD Moment = -6.38 FT-K or <sup>3/4 Point</sup> 15.12 FT-K

RAISE LOAD = 12.98 K or 12.76 FT-K

## CALCULATION SHEET

PAGE 19 OF     PROJECT NO. 19116CLIENT NY DOT SUBJECT CEMETARY POOPPrepared By JM Date 9/98PROJECT LIE UNDERPASSReviewed By WV Date 9/98Approved By      Date     

LIVE LOAD @ 3/8 POINT

$$M_{LL(\text{NEGATIVE})} = 16^4 \times 1.20 \times (-4.07 + 1.03) \times 0.174 = 17.04 \text{ FT-K}$$

$$P_{LL(\text{NEGATIVE})} = 16^4 \times 1.20 \times (-0.55 + 0.22) \times 0.174 = 2.57 \text{ K}$$

$$\text{DEAD LOAD MOMENT} = 20.09 \text{ FT-K}$$

$$\text{AXIAL LOAD} = 12.36$$

$$M_{LL(\text{POSITIVE})} = 16^4 \times 1.20 \times (1.05 + 0.80) \times 0.174 = 6.18 \text{ FT-K}$$

$$P_{LL(\text{POSITIVE})} = 16^4 \times 1.20 \times (0.56 + 0.23) \times 0.174 = 2.64 \text{ FT-K}$$

## CALCULATION SHEET

PAGE 20 OF     PROJECT NO. 19116CLIENT NY DOTSUBJECT CEMETARY ROADPrepared By JM Date 9/98PROJECT LIEUNDER PASSReviewed By WV Date 9/98Approved By      Date     

LIVE LOAD ANALYSIS @ CROWN

$$M_{LL} = 16^k \times 1.20(0.40 + 3.89) + 4^k(0.40) \times 1.10 = 84.1$$

$$D.F. = 0.174$$

$$M_{LL} = 84.1 \times 0.174 = 14.63 \text{ FT-K}$$

AXIAL LOAD

$$P = 16^k \times 1.2(0.62 + 0.41) + 4^k \times 1.10(0.41) = 21.58$$

$$P_{LL} = 21.58 \times 0.174 = 3.75^k$$

$$\text{DEAD LOAD MOMENT} = 25.28 \text{ FT-K} \quad \text{AXIAL LOAD} = 12.26^k$$

CALCULATION SHEET

PAGE 21 OF     

PROJECT NO. 19116

CLIENT NY DOT

SUBJECT Concrete Road

Prepared By JM Date 9/99

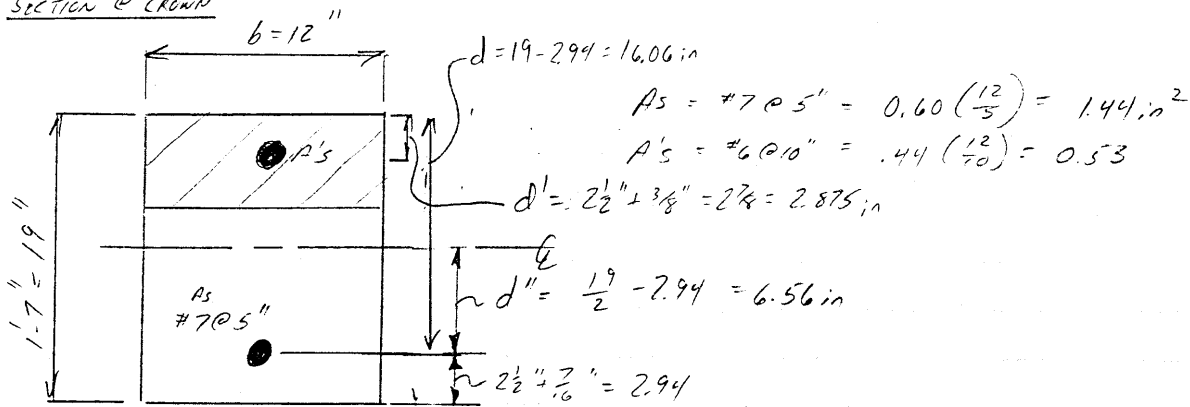
PROJECT LIE

UNASSUMED

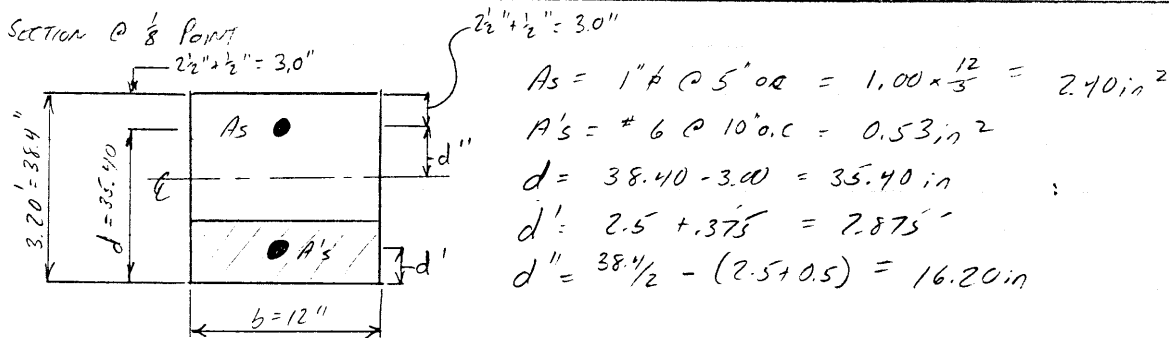
Reviewed By WN Date 9/99

Approved By      Date     

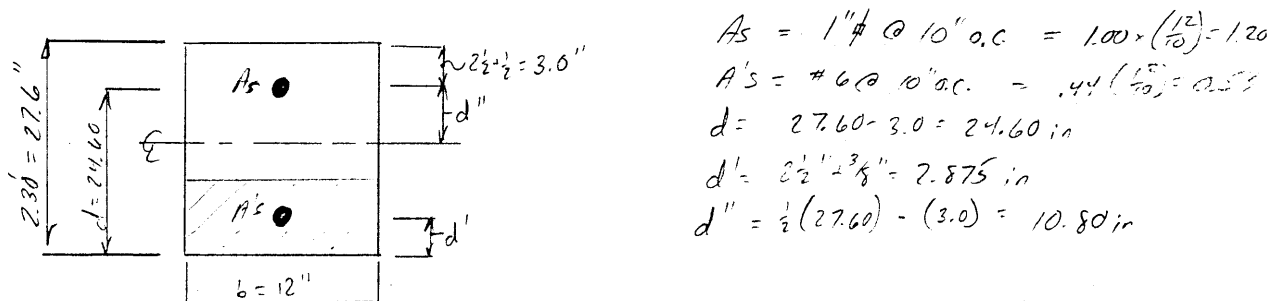
SECTION @ CROWN



SECTION @ 1/8 POINT



SECTION @ 1/4 POINT





CALCULATION SHEET

PAGE 22 OF     

PROJECT NO. 19116

CLIENT IND DOT SUBJECT CONCRETE ROAD

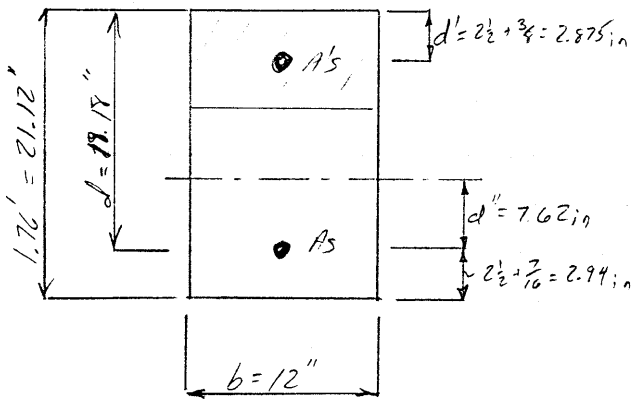
Prepared By JM Date 9/98

PROJECT LIC UNDERPASS

Reviewed By WV Date 9/98

Approved By      Date     

SECTION @ 3/8 POINT (POSITIVE MOMENT)



$$A_s = \#7 @ 5" O.C = 0.60 \left(\frac{12}{5}\right) = 1.44 in^2$$

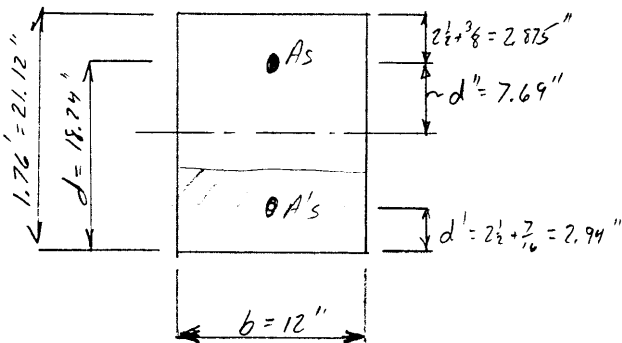
$$A'_s = \#6 @ 10" O.C = 0.44 \left(\frac{12}{10}\right) = 0.53 in^2$$

$$d = 21.12 - 2.94 = 18.18 in$$

$$d' = 2\frac{1}{2} + \frac{3}{8} = 2.875$$

$$d'' = \frac{1}{2}(21.12) - 2.94 = 7.62$$

SECTION @ 3/8 POINT (NEGATIVE MOMENT)



$$A_s = \#6 @ 10" O.C = 0.53 in^2$$

$$A'_s = \#7 @ 5" O.C = 1.44 in^2$$

$$d = 21.12 - 2.88 = 18.24 in$$

$$d' = 2\frac{1}{2} + \frac{3}{8} = 2.875$$

$$d'' = \frac{1}{2}(21.12) - 2.88 = 7.69 in$$

CALCULATION SHEET

PAGE 23 OF     

PROJECT NO. 19116

CLIENT NY DOT SUBJECT CEMETARY ROAD

Prepared By JM Date 9/98

PROJECT LIE UNDER PACE

Reviewed By JV Date 9/98

Approved By      Date     

RATING @ Abt Force

$$\frac{18,000 - 13144}{4507} = 1.077 \quad \begin{array}{l} \times 36 \text{ Tons} = 38.8 \text{ Tons} \\ \times 1520 = 1521.5 \end{array} \quad \text{INVENTORY}$$

$$\frac{25,000 - 13144}{4507} = 2.631 \quad \begin{array}{l} \times 36 \text{ Tons} = 94.7 \text{ Tons} \\ \times 1520 = 1520.6 \end{array} \quad \text{OPERATING}$$

RATING @ 1/8th Point

$$\frac{18,000 - 7087}{4515} = 2.417 \quad \begin{array}{l} \times 36 \text{ Tons} = 87.0 \text{ Tons} \\ \times 1520 = 1520.3 \end{array} \quad \text{INVENTORY}$$

$$\frac{25,000 - 7087}{4515} = 3.967 \quad \begin{array}{l} \times 36 \text{ Tons} = 142.8 \text{ Tons} \\ \times 1520 = 1520.3 \end{array} \quad \text{OPERATING}$$

RATING @ 1/4 Point

$$\frac{18,000 - 2249}{6825} = 2.308 \quad \begin{array}{l} \times 36 \text{ Tons} = 83.1 \text{ Tons} \\ \times 1520 = 1520.2 \end{array} \quad \text{INVENTORY}$$

$$\frac{25,000 - 2249}{6825} = 3.333 \quad \begin{array}{l} \times 36 \text{ Tons} = 120.0 \text{ Tons} \\ \times 1520 = 1520.7 \end{array} \quad \text{OPERATING}$$

CALCULATION SHEET

PAGE 24 OF     

PROJECT NO. 19116

CLIENT NV 607

SUBJECT CEMENTARY ROAD

Prepared By JM Date 9/98

PROJECT     

INDUSTRIAL

Reviewed By WV Date 9/98

Approved By      Date     

PA-116 @ 3/8th Point (Positive Percent Analysis)

$$\frac{18,000 - 6389}{7917} = 1.467 \times 36 \text{ Tons} = 52.8 \text{ Tons} \quad \text{INVENTORY}$$

$$\hspace{12em} \times \text{HS20} = \text{HS } 29.3$$

$$\frac{25,000 - 6389}{7917} = 2.351 \times 36 \text{ Tons} = 84.6 \text{ Tons} \quad \text{OPERATING}$$

$$\hspace{12em} \times \text{HS20} = \text{HS } 47.0$$

RATING @ 3/8th Point (Negative Percent Analysis)

$$\frac{18,000 - (-15830)}{5921} =$$

by inspection does not control 3/8th point

$$\frac{25,000 - (-15830)}{5921} =$$

Rating @ Crown

$$\frac{18,000 - 10808}{7340} = 0.980 \times 36 \text{ Tons} = 35.3 \text{ Tons} \quad \text{INVENTORY}$$

$$\hspace{12em} \times \text{HS20} = \text{HS } 19.6 \text{ Tons}$$

$$\frac{25,000 - 10808}{7340} = 1.934 \times 36 \text{ Tons} = 69.6 \text{ Tons} \quad \text{OPERATING}$$

$$\hspace{12em} \times \text{HS20} = \text{HS } 38.7$$

WV 9/98

Reinforced concrete analysis/ w compression reinforcement

Combined bending and axial load

Member stresses at face of abutment

due to dead load

$f_c = 2500 \text{ psi}$  ✓  
 $n = 12$  ✓  
 $b = 12.00 \text{ in}$  ✓  
 $d = 51.00 \text{ in}$  ✓  
 $d' = 2.88 \text{ in}$  ✓  
 $d'' = 24.00 \text{ in}$  ✓  
 $M = 152.65 \text{ ft-k}$  ✓  
 $P = 17.13 \text{ kip}$  ✓  
 $A_s = 2.40 \text{ in}^2$  ✓  
 $A's = 0.53 \text{ in}^2$  ✓

$$e = 12M / N + d'' = 12 \times 152.7 / 17.1 + 24.00 = 130.9$$

$$e/d = 130.9 / 51.00 = 2.57$$

Assume  $j = 0.903$

$$i = 1 / 1 - jd/e = 1 / 1 - 0.903 \times 51.00 / 130.9 = 1.54$$

$$m = n A_s i / b d + (2n - 1) A's / b d =$$

$$m = ( 12 \times 2.4 \times 1.54 / 12 \times 51 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 51$$

$$m = ( 0.073 ) + ( 0.020 ) = 0.093$$

$$q = n A_s i / b d + (2n - 1) A's / b d \times d' / d$$

$$q = ( 12 \times 2.4 \times 1.54 / 12 \times 51 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 51 \times 2.88 / 51$$

$$q = ( 0.073 ) + ( 0.020 ) \times ( 0.056 ) = 0.074$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.302$$

$$z = \frac{1}{6} + \frac{(2n-1) A's}{k b d} \times \frac{d'}{k d} \times \frac{(1 - d')}{k d} = \frac{0.177}{0.554} = 0.319$$

$$\frac{1}{2} + \frac{(2n-1) A's}{k b d} \times \frac{(1 - d')}{k d}$$

$$j = 1 - z k = 1 - 0.319 \times 0.302 = 0.903 \text{ check}$$

$$i = (1 / (1 - (j d / e))) = 1.543$$

$$f_s = (P / j A_s i) \times e/d = 13144 \text{ psi}$$

$$f_c = f_s / n \times (k / (1 - k)) = 475 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement  
 Combined bending and axial load  
 Member stresses at face of abutment  
 due to live load

- $f_c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 51.00$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 24.00$  in ✓
- $M = 49.66$  ft-k ✓
- $P = 4.28$  kip ✓
- $A_s = 2.40$  in<sup>2</sup> ✓
- $A's = 0.53$  in<sup>2</sup> ✓

$$e = 12M / N + d'' = 12 \times 49.7 / 4.3 + 24.00 = 163.2$$

$$e/d = 163.2 / 51.00 = 3.20$$

Assume  $j = 0.907$

$$i = 1 / 1 - j d/e = 1 / 1 - 0.907 \times 51.00 / 163.2 = 1.40$$

$$m = n A_s i / b d + (2n - 1) A's / b d =$$

$$m = ( 12 \times 2.4 \times 1.4 / 12 \times 51 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 51$$

$$m = ( 0.066 ) + ( 0.020 ) = 0.086$$

$$q = n A_s i / b d + (2n - 1) A's / b d \times d' / d$$

$$q = ( 12 \times 2.4 \times 1.4 / 12 \times 51 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 51 \times 2.88 / 51$$

$$q = ( 0.066 ) + ( 0.020 ) \times ( 0.056 ) = 0.067$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.290$$

$$z = \frac{1}{6} + \frac{(2n-1) A's}{k b d} \times \frac{d'}{k d} \times \frac{(1 - d')}{k d} = \frac{0.177}{0.555} = 0.319$$

$$\frac{1}{2} + \frac{(2n-1) A's}{k b d} \times \frac{(1 - d')}{k d}$$

$$j = 1 - z k = 1 - 0.319 \times 0.290 = 0.907 \text{ check}$$

$$i = (1/(1-(j d/e))) = 1.396$$

$$f_s = (P/j A_s i) e/d = 4507 \text{ psi}$$

$$f_c = f_s / n (k/(1-k)) = 153 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement

$f_c = 2500$  psi ✓  
 $n = 12$  ✓  
 $b = 12.00$  in ✓  
 $d = 35.40$  in ✓  
 $d' = 2.88$  in ✓  
 $d'' = 16.20$  in ✓  
 $M = 61.35$  ft-k ✓  
 $P = 13.73$  kip ✓  
 $A_s = 2.40$  in<sup>2</sup> ✓  
 $A's = 0.53$  in<sup>2</sup> ✓

Combined bending and axial load  
 Member stresses at 1 / 8 point  
 due to dead load

$$e = 12M / N + d'' = 12 \times 61.4 / 13.7 + 16.20 = 69.8$$

$$e / d = 69.8 / 35.40 = 1.97$$

Assume  $j = 0.881$

$$i = 1 / 1 - j d / e = 1 / 1 - 0.881 \times 35.40 / 69.8 = 1.81$$

$$m = n A_s i / b d + (2n - 1) A's / b d =$$

$$m = ( 12 \times 2.4 \times 1.81 / 12 \times 35.4 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 35.4$$

$$m = ( 0.123 ) + ( 0.029 ) = 0.151$$

$$q = n A_s i / b d + (2n - 1) A's / b d \times d' / d$$

$$q = ( 12 \times 2.4 \times 1.81 / 12 \times 35.4 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 35.4 \times 2.88 / 35.4$$

$$q = ( 0.123 ) + ( 0.029 ) \times ( 0.081 ) = 0.125$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.371$$

$$z = \frac{1}{6} + \frac{(2n-1)A's}{k b d} \times \frac{d'}{k d} \times \frac{(1-d')}{k d} = \frac{0.180}{0.560} = 0.321$$

$$\frac{1}{2} + \frac{(2n-1)A's}{k b d} \times \frac{(1-d')}{k d}$$

$$j = 1 - z k = 1 - 0.321 \times 0.371 = 0.881 \text{ check}$$

$$i = (1 / (1 - (j d / e))) = 1.807$$

$$f_s = (P / j A_s i) \times e / d = 7087 \text{ psi}$$

$$f_c = f_s / n \times (k / (1 - k)) = 348 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement

- $f'_c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 35.40$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 16.20$  in ✓
- $M = 33.73$  ft-k ✓
- $P = 4.04$  kip ✓
- $A_s = 2.40$  in<sup>2</sup> ✓
- $A'_s = 0.53$  in<sup>2</sup> ✓

Combined bending and axial load  
 Member stresses at 1 / 8 point  
 due to live load

$$e = 12M / N + d'' = 12 \times 33.7 / 4.0 + 16.20 = 116.4$$

$$e / d = 116.4 / 35.40 = 3.29$$

Assume  $j = 0.893$

$$i = 1 / 1 - jd/e = 1 / 1 - 0.893 \times 35.40 / 116.4 = 1.37$$

$$m = n A_s i / b d + (2n - 1) A'_s / b d =$$

$$m = ( 12 \times 2.4 \times 1.37 / 12 \times 35.4 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 35.4$$

$$m = ( 0.093 ) + ( 0.029 ) = 0.122$$

$$q = n A_s i / b d + (2n - 1) A'_s / b d \times d' / d$$

$$q = ( 12 \times 2.4 \times 1.37 / 12 \times 35.4 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 35.4 \times 2.88 / 35.4$$

$$q = ( 0.093 ) + ( 0.029 ) \times ( 0.081 ) = 0.095$$

$$k = ( m^2 + 2 q^{1/2} - m )^{0.332}$$

$$z = \frac{1}{6} + \frac{(2n-1) A'_s}{k b d} \times \frac{d'}{k d} \times \frac{(1 - d')}{k d} = \frac{0.183}{0.565} = 0.323$$

$$\frac{1}{2} + \frac{(2n-1) A'_s}{k b d} \times \frac{(1 - d')}{k d}$$

$$j = 1 - z k = 1 - 0.323 \times 0.332 = 0.893 \text{ check}$$

$$i = (1 / (1 - (j d / e))) = 1.373$$

$$f_s = (P / j A_s i) e / d = 4515 \text{ psi}$$

$$f_c = f_s / n (k / (1 - k)) = 187 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement  
 Combined bending and axial load  
 Member stresses at 1/4 point  
 due to dead load

W.V 9/198

- $f_c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 24.60$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 10.80$  in ✓
- $M = 15.12$  ft-k ✓
- $P = 12.76$  kip ✓
- $A_s = 1.20$  in<sup>2</sup> ✓
- $A's = 0.53$  in<sup>2</sup> ✓

$$e = 12M / N + d'' = 12 \times 15.1 / 12.8 + 10.80 = 25.0$$

$$e/d = 25.0 / 24.60 = 1.02$$

Assume  $j = 0.839$

$$i = 1 / 1 - j d/e = 1 / 1 - 0.839 \times 24.60 / 25.0 = 5.71$$

$$m = n A_s i / b d + (2n - 1) A's / b d =$$

$$m = ( 12 \times 1.2 \times 5.71 / 12 \times 24.6 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 24.6$$

$$m = ( 0.279 ) + ( 0.041 ) = 0.320$$

$$q = n A_s i / b d + (2n - 1) A's / b d \times d' / d$$

$$q = ( 12 \times 1.2 \times 5.71 / 12 \times 24.6 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 24.6 \times 2.88 / 24.6$$

$$q = ( 0.279 ) + ( 0.041 ) \times ( 0.117 ) = 0.283$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.498$$

$$z = \frac{\frac{1}{6} + \frac{(2n-1) A's}{k b d} \times \frac{d'}{k d} \times (1 - \frac{d'}{k d}}{\frac{1}{2} + \frac{(2n-1) A's}{k b d} \times (1 - \frac{d'}{k d})}} = \frac{0.182}{0.563} = 0.322$$

$$j = 1 - z k = 1 - 0.322 \times 0.498 = 0.839 \text{ check}$$

$$i = (1 / (1 - j d / e)) = 5.728$$

$$f_s = (P / j A_s i) \times e/d = 2249 \text{ psi}$$

$$f_c = f_s / n \times (k / (1 - k)) = 186 \text{ psi}$$



Reinforced concrete analysis/ w compression reinforcement

W.V. 9/98

- $f'c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 24.60$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 10.80$  in ✓
- $M = 18.17$  ft-k ✓
- $P = 3.21$  kip ✓
- $A_s = 1.20$  in<sup>2</sup> ✓
- $A's = 0.53$  in<sup>2</sup> ✓

Combined bending and axial load  
 Member stresses at 1 / 4 point  
 due to live load

$$e = 12M / N + d'' = 12 \times 18.2 / 3.2 + 10.80 = 78.7$$

$$e/d = 78.7 / 24.60 = 3.20$$

Assume  $j = 0.901$

$$i = 1 / 1 - j d/e = 1 / 1 - 0.901 \times 24.60 / 78.7 = 1.39$$

$$m = n A_s i / b d + (2n - 1) A's / b d =$$

$$m = ( 12 \times 1.2 \times 1.39 / 12 \times 24.6 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 24.6$$

$$m = ( 0.068 ) + ( 0.041 ) = 0.109$$

$$q = n A_s i / b d + (2n - 1) A's / b d \times d' / d$$

$$q = ( 12 \times 1.2 \times 1.39 / 12 \times 24.6 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 24.6 \times 2.88 / 24.6$$

$$q = ( 0.068 ) + ( 0.041 ) \times ( 0.117 ) = 0.073$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.288$$

$$z = \frac{1}{6} + \frac{(2n-1)A's}{k b d} \times \frac{d'}{k d} \times \frac{(1-d')}{k d} = \frac{0.201}{0.585} = 0.344$$

$$\frac{1}{2} + \frac{(2n-1)A's}{k b d} \times \frac{(1-d')}{k d}$$

$$j = 1 - z k = 1 - 0.344 \times 0.288 = 0.901 \text{ check}$$

$$i = (1/(1-(j d/e))) = 1.392$$

$$f_s = (P/j A_s i) e/d = 6825 \text{ psi}$$

$$f_c = f_s / n (k/(1-k)) = 230 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement

- f<sub>c</sub> = 2500 psi ✓
- n = 12 ✓
- b = 12.00 in ✓
- d = 18.18 in ✓
- d' = 2.88 in ✓
- d'' = 7.62 in ✓
- M = 20.09 ft-k ✓
- P = 12.36 kip ✓
- A<sub>s</sub> = 1.44 in<sup>2</sup> ✓
- A'<sub>s</sub> = 0.53 in<sup>2</sup> ✓

Combined bending and axial load  
 Member stresses at 3/8 point  
 due to dead load

*Positive Moment Analysis  
 (Tension on bottom)*

$$e = 12M / N + d'' = 12 \times 20.1 / 12.4 + 7.62 = 27.1$$

$$e/d = 27.1 / 18.18 = 1.49$$

Assume j = 0.855

$$i = 1 / (1 - jd/e) = 1 / (1 - 0.855 \times 18.18 / 27.1) = 2.34$$

$$m = n A_s i / b d + (2n - 1) A'_s / b d =$$

$$m = ( 12 \times 1.44 \times 2.34 / 12 \times 18.2 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 18.2$$

$$m = ( 0.186 ) + ( 0.056 ) = 0.241$$

$$q = n A_s i / b d + (2n - 1) A'_s / b d \times d' / d$$

$$q = ( 12 \times 1.44 \times 2.34 / 12 \times 18.2 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 18.2 \times 2.88 / 18.2$$

$$q = ( 0.186 ) + ( 0.056 ) \times ( 0.158 ) = 0.194$$

$$k = ( m^2 + 2q )^{1/2} - m = 0.427$$

$$z = \frac{1}{6} + \frac{(2n-1) A'_s}{k b d} \times \frac{d'}{k d} \times \frac{(1-d')}{k d} = \frac{0.197}{0.582} = 0.339$$

$$\frac{1}{2} + \frac{(2n-1) A'_s}{k b d} \times \frac{(1-d')}{k d}$$

$$j = 1 - z k = 1 - 0.339 \times 0.427 = 0.855 \text{ check}$$

$$i = (1/(1-(j d/e))) = 2.343$$

$$f_s = (P/j A_s i) e/d = 6389 \text{ psi}$$

$$f_c = f_s / n (k/(1-k)) = 397 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement

W.V. 9198

- $f_c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 18.18$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 7.62$  in ✓
- $M = 17.04$  ft-k ✓
- $P = 2.57$  kip ✓
- $A_s = 1.44$  in<sup>2</sup> ✓
- $A'_s = 0.53$  in<sup>2</sup> ✓

Combined bending and axial load

Member stresses at 3 / 8 point  
 due to live load

Positive Moment Analysis  
 (Tension on Bottom)

$$e = 12M / N + d'' = 12 \times 17.0 / 2.6 + 7.62 = 87.2$$

$$e/d = 87.2 / 18.18 = 4.80$$

Assume  $j = 0.882$

$$i = 1 / 1 - j d/e = 1 / 1 - 0.882 \times 18.18 / 87.2 = 1.23$$

$$m = n A_s i / b d + (2n - 1) A'_s / b d =$$

$$m = ( 12 \times 1.44 \times 1.23 / 12 \times 18.2 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 18.2$$

$$m = ( 0.097 ) + ( 0.056 ) = 0.153$$

$$q = n A_s i / b d + (2n - 1) A'_s / b d \times d' / d$$

$$q = ( 12 \times 1.44 \times 1.23 / 12 \times 18.2 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 18.2 \times 2.88 / 18.2$$

$$q = ( 0.097 ) + ( 0.056 ) \times ( 0.158 ) = 0.106$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.332$$

$$z = \frac{1}{6} + \frac{(2n-1) A'_s}{k b d} \times \frac{d'}{k d} \times \frac{(1 - d')}{k d}}{2 + \frac{(2n-1) A'_s}{k b d} \times \frac{(1 - d')}{k d}} = \frac{0.209}{0.588} = 0.355$$

$$j = 1 - z k = 1 - 0.355 \times 0.332 = 0.882 \text{ check}$$

$$i = (1 / (1 - (j d / e))) = 1.225$$

$$f_s = (P / j A_s i) \times e/d = 7917 \text{ psi}$$

$$f_c = f_s / n \times (k / (1 - k)) = 328 \text{ psi}$$

Reinforced concrete analysis/ w compression reinforcement

W.V. 9198

- $f_c = 2500$  psi ✓
- $n = 12$  ✓
- $b = 12.00$  in ✓
- $d = 16.06$  in ✓
- $d' = 2.88$  in ✓
- $d'' = 6.56$  in ✓
- $M = 25.28$  ft-k ✓
- $P = 12.26$  kip ✓
- $A_s = 1.44$  in<sup>2</sup> ✓
- $A'_s = 0.53$  in<sup>2</sup> ✓

Combined bending and axial load  
 Member stresses at crown  
 due to dead load

$$e = 12M / N + d'' = 12 \times 25.3 / 12.3 + 6.56 = 31.3$$

$$e/d = 31.3 / 16.06 = 1.95$$

Assume  $j = 0.859$

$$i = 1 / 1 - j d/e = 1 / 1 - 0.859 \times 16.06 / 31.3 = 1.79$$

$$m = n A_s i / b d + (2n - 1) A'_s / b d =$$

$$m = ( 12 \times 1.44 \times 1.79 / 12 \times 16.1 ) + ( 2 \times 12 - 1 ) \times 0.53 / 12 \times 16.1$$

$$m = ( 0.160 ) + ( 0.063 ) = 0.224$$

$$q = n A_s i / b d + (2n - 1) A'_s / b d \times d'/d$$

$$q = ( 12 \times 1.44 \times 1.79 / 12 \times 16.1 ) + ( 2 \times 0.53 - 1 ) \times 0.53 / 12 \times 16.1 \times 2.88 / 16.1$$

$$q = ( 0.160 ) + ( 0.063 ) \times ( 0.179 ) = 0.172$$

$$k = ( m^2 + 2 q^{1/2} - m ) / 0.404$$

$$z = \frac{1}{6} + \frac{(2n-1) A'_s}{k b d} \times \frac{d'}{k d} \times \frac{(1 - d')}{k d} = \frac{0.205}{0.587} = 0.350$$

$$\frac{1}{2} + \frac{(2n-1) A'_s}{k b d} \times \frac{(1 - d')}{k d}$$

$$j = 1 - z k = 1 - 0.350 \times 0.404 = 0.859 \text{ check}$$

$$i = (1/(1-(j d/e))) = 1.788$$

$$f_s = (P/j A_s i) e/d = 10808 \text{ psi}$$

$$f_c = f_s / n (k/(1-k)) = 609 \text{ psi}$$

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*****
*
*           S T A A D - III
*           Revision 22.3a
*           Proprietary Program of
*           Research Engineers, Inc.
*           Date=   SEP  2, 1998
*           Time=   11: 3:50
*
*           USER ID: Rust E & I
*****

```

1. STAAD PLANE CEMETERY ROAD UNDERPASS
2. INPUT WIDTH 72
3. UNIT FEET KIP
4. JOINT COORDINATES
5. 1 0.0 0.0; 2 -0.23 3.33; 3 -0.43 6.33; 4 -0.63 9.33
6. 5 -0.83 12.33; 6 -0.95 14.06; 7 1.33 14.58; 8 4.75 15.21
7. 9 8.16 15.75; 10 11.58 16.20; 11 14.99 16.47; 12 18.40 16.88
8. 13 21.82 16.95; 14 25.23 17.09; 15 28.65 17.16; 16 32.06 17.09
9. 17 35.48 16.95; 18 38.89 16.88; 19 42.30 16.47; 20 45.72 16.20
10. 21 49.13 15.75; 22 52.55 15.21; 23 55.96 14.58; 24 58.25 14.06
11. 25 58.13 12.33; 26 57.93 9.33; 27 57.72 6.33; 28 57.52 3.33
12. 29 57.29 0.0
13. MEMBER INCIDENCE
14. 1 1 2; 2 2 3; 3 3 4; 4 4 5; 5 5 6; 6 6 7; 7 7 8; 8 8 9
15. 9 9 10; 10 10 11; 11 11 12; 12 12 13; 13 13 14; 14 14 15
16. 15 15 16; 16 16 17; 17 17 18; 18 18 19; 19 19 20; 20 20 21
17. 21 21 22; 22 22 23; 23 23 24; 24 24 25; 25 25 26; 26 26 27
18. 27 27 28; 28 28 29
19. UNIT INCHES KIP
20. MEMBER PROPERTY AMERICAN
21. 1 28 PRI AX 410. IX 40538.0 IZ 162153.0
22. 2 27 PRI AX 471. IX 61099.0 IZ 244396.0
23. 3 26 PRI AX 529. IX 86567.0 IZ 346267.0
24. 4 25 PRI AX 588. IX 118207.0 IZ 472829.0
25. 5 24 PRI AX 626. IX 142347.0 IZ 569389.0
26. 6 23 PRI AX 642. IX 153339.0 IZ 613357.0
27. 7 22 PRI AX 598. IX 126141.0 IZ 504566.0
28. 8 21 PRI AX 504. IX 75721.0 IZ 302884.0
29. 9 20 PRI AX 425. IX 45318.0 IZ 181271.0
30. 10 19 PRI AX 360. IX 27518.0 IZ 110074.0
31. 11 18 PRI AX 308. IX 17219.0 IZ 68876.0
32. 12 17 PRI AX 269. IX 11417.0 IZ 45668.0
33. 13 16 PRI AX 244. IX 8452.0 IZ 33808.0
34. 14 15 PRI AX 231. IX 7150.0 IZ 28599.0
35. CONSTANT
36. E CONCRETE ALL
37. UNIT FEET POUND
38. CONSTANT
39. DENSITY CONCRETE ALL
40. UNIT FEET KIP
41. SUPPORT

42. 1 29 PINNED  
43. UNIT FEET POUND  
44. DEFINE MOVING LOAD  
45. TYPE 1 LOAD 8000.0 32000.0 32000.0 DISTANCE 14.0 14.0  
46. LOAD 1  
47. SELFWEIGHT Y -1.  
48. LOAD 2  
49. MEMBER LOAD  
50. 6 TO 23 UNI GY -162.0  
51. LOAD 3  
52. MEMBER LOAD  
53. 1 TRAP GX 358.2 298.8  
54. 2 TRAP GX 298.8 239.4  
55. 3 TRAP GX 239.4 180.0  
56. 4 TRAP GX 180.0 120.6  
57. 5 TRAP GX 120.6 61.2  
58. LOAD 4  
59. MEMBER LOAD  
60. 28 TRAP GX -358.2 -298.8  
61. 7 TRAP GX -298.8 -239.4  
62. 26 TRAP GX -239.4 -180.0  
63. 25 TRAP GX -180.0 -120.6  
64. 24 TRAP GX -120.6 61.2  
65. LOAD 5  
66. MEMBER LOAD  
67. 6 TRAP GY -322.5 -319.5  
68. 7 TRAP GY -319.5 -262.5  
69. 8 TRAP GY -262.5 -213.0  
70. 9 TRAP GY -213.0 -166.5  
71. 10 TRAP GY -166.5 -126.0  
72. 11 TRAP GY -126.0 -91.5  
73. 12 TRAP GY -91.5 -63.0  
74. 13 TRAP GY -63.0 -39.0  
75. 14 TRAP GY -39.0 -21.0  
76. 15 TRAP GY -21.0 -9.0  
77. 16 TRAP GY -9.0 1.5  
78. 17 TRAP GY 1.5 -0.0  
79. 18 TRAP GY -0.0 -4.5  
80. 19 TRAP GY -4.5 -13.5  
81. 20 TRAP GY -13.5 -28.5  
82. 21 TRAP GY -28.5 -48.0  
83. 22 TRAP GY -48.0 -73.5  
84. 23 TRAP GY -73.5 -75.0  
85. LOAD 6  
86. MEMBER LOAD  
87. 6 TO 23 UNI GY -28.9  
88. LOAD COMB 7 TOTAL DEAD  
89. 1 1. 2 1. 3 1. 4 1. 5 1. 6 1.  
90. LOAD COMB 8 TOTAL DEAD-NO EARTH  
91. 1 1. 2 1. 3 2. 4 2. 5 1. 6 1.  
92. \*LOAD GENERATION 20  
93. \*TYPE 1 -0.95 14.06 0.0 XINC 2.73 YRA 5.0  
94. PERFORM ANALYSIS

P R O B L E M   S T A T I S T I C S  
-----

NUMBER OF JOINTS/MEMBER+ELEMENTS/SUPPORTS =    29/   28/    2  
ORIGINAL/FINAL BAND-WIDTH =    1/    1  
TOTAL PRIMARY LOAD CASES =    6, TOTAL DEGREES OF FREEDOM =    83  
SIZE OF STIFFNESS MATRIX =    498 DOUBLE PREC. WORDS  
REQRD/AVAIL. DISK SPACE = 12.05/ 369.3 MB, EXMEM = 1964.9 MB

++ Processing Element Stiffness Matrix.                    11: 3:51  
++ Processing Global Stiffness Matrix.                    11: 3:52  
++ Processing Triangular Factorization.                    11: 3:52  
++ Calculating Joint Displacements.                        11: 3:52  
++ Calculating Member Forces.                              11: 3:52

95. PRINT MEMBER FORCES ALL

40

## MEMBER END FORCES      STRUCTURE TYPE = PLANE

-----  
ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
1	1	1	19129.59	7920.21	.00	.00	.00	-.01
		2	-17707.50	-7821.99	.00	.00	.00	26273.19
	2	1	4585.54	3723.80	.00	.00	.00	.01
		2	-4585.54	-3723.80	.00	.00	.00	12429.79
	3	1	-122.96	-2637.84	.00	.00	.00	.00
		2	198.51	1543.93	.00	.00	.00	-6924.17
	4	1	398.83	927.28	.00	.00	.00	-.01
		2	-398.83	-927.28	.00	.00	.00	3095.20
	5	1	3944.03	1456.12	.00	.00	.00	.00
		2	-3944.03	-1456.12	.00	.00	.00	4860.44
	6	1	818.04	664.31	.00	.00	.00	.00
		2	-818.04	-664.31	.00	.00	.00	2217.41
	7	1	28753.08	12053.88	.00	.00	.00	.00
		2	-27255.43	-13049.56	.00	.00	.00	41951.87
	8	1	29028.95	10343.32	.00	.00	.00	-.01
		2	-27455.74	-12432.91	.00	.00	.00	38122.90
2	1	2	17726.15	7779.63	.00	.00	.00	-26273.18
		3	-16254.37	-7681.51	.00	.00	.00	49516.36
	2	2	4594.43	3712.81	.00	.00	.00	-12429.76
		3	-4594.43	-3712.81	.00	.00	.00	23593.00
	3	2	-202.20	-1543.47	.00	.00	.00	6924.16
		3	256.02	736.17	.00	.00	.00	-10306.48
	4	2	401.05	926.32	.00	.00	.00	-3095.18
		3	-401.05	-926.32	.00	.00	.00	5880.33
	5	2	3947.50	1446.69	.00	.00	.00	-4860.41
		3	-3947.50	-1446.69	.00	.00	.00	9210.16
	6	2	819.62	662.35	.00	.00	.00	-2217.42
		3	-819.62	-662.35	.00	.00	.00	4208.87
	7	2	27286.55	12984.34	.00	.00	.00	-41951.80
		3	-25760.95	-13693.52	.00	.00	.00	82102.23
	8	2	27485.40	12367.19	.00	.00	.00	-38122.82
		3	-25905.97	-13883.68	.00	.00	.00	77676.07
3	1	3	16254.37	7681.58	.00	.00	.00	-49516.34
		4	-14601.35	-7571.38	.00	.00	.00	72446.38
	2	3	4594.43	3712.87	.00	.00	.00	-23592.92
		4	-4594.43	-3712.87	.00	.00	.00	34756.25
	3	3	-256.02	-736.12	.00	.00	.00	10306.58
		4	297.96	107.02	.00	.00	.00	-11529.37
	4	3	401.05	926.31	.00	.00	.00	-5880.30
		4	-401.05	-926.31	.00	.00	.00	8665.55
	5	3	3947.50	1446.72	.00	.00	.00	-9210.10
		4	-3947.50	-1446.72	.00	.00	.00	13559.89



41

MEMBER END FORCES      STRUCTURE TYPE = PLANE

-----  
 ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
6		3	819.62	662.35	.00	.00	.00	-4208.86
		4	-819.62	-662.35	.00	.00	.00	6200.33
7		3	25760.96	13693.71	.00	.00	.00	-82101.94
		4	-24065.99	-14212.61	.00	.00	.00	124099.00
8		3	25905.97	13883.90	.00	.00	.00	-77675.64
		4	-24169.07	-15031.90	.00	.00	.00	121235.20
4	1	4	14601.34	7571.44	.00	.00	.00	-72446.26
		5	-12763.96	-7448.95	.00	.00	.00	95026.59
2	4	4	4594.43	3712.85	.00	.00	.00	-34756.16
		5	-4594.43	-3712.85	.00	.00	.00	45919.41
3	4	4	-297.96	-107.11	.00	.00	.00	11529.34
		5	328.02	-343.79	.00	.00	.00	-11128.84
4	4	4	401.05	926.32	.00	.00	.00	-8665.37
		5	-401.05	-926.32	.00	.00	.00	11450.54
5	4	4	3947.50	1446.61	.00	.00	.00	-13559.80
		5	-3947.50	-1446.61	.00	.00	.00	17909.49
6	4	4	819.62	662.36	.00	.00	.00	-6200.32
		5	-819.62	-662.36	.00	.00	.00	8191.79
7	4	4	24065.99	14212.48	.00	.00	.00	-124098.60
		5	-22198.55	-14540.89	.00	.00	.00	167369.00
8	4	4	24169.07	15031.70	.00	.00	.00	-121234.60
		5	-22271.57	-15811.01	.00	.00	.00	167690.70
5	1	5	12743.91	7483.62	.00	.00	.00	-95026.55
		6	-11615.88	-7405.37	.00	.00	.00	107935.50
2	5	5	4584.45	3725.03	.00	.00	.00	-45919.25
		6	-4584.45	-3725.03	.00	.00	.00	52379.42
3	5	5	-328.94	341.94	.00	.00	.00	11128.34
		6	339.85	-499.20	.00	.00	.00	-10383.55
4	5	5	398.56	927.32	.00	.00	.00	-11450.83
		6	-398.56	-927.32	.00	.00	.00	13059.45
5	5	5	3943.60	1457.08	.00	.00	.00	-17909.55
		6	-3943.60	-1457.08	.00	.00	.00	20437.08
6	5	5	817.84	664.53	.00	.00	.00	-8191.81
		6	-817.84	-664.53	.00	.00	.00	9344.20
7	5	5	22159.42	14599.53	.00	.00	.00	-167369.60
		6	-21020.48	-14678.54	.00	.00	.00	192772.10
8	5	5	22229.04	15868.79	.00	.00	.00	-167692.10
		6	-21079.19	-16105.06	.00	.00	.00	195448.00
6	1	6	9109.11	10333.66	.00	.00	.00	107935.30
		7	-8761.38	-8809.01	.00	.00	.00	-85552.54
2	6	6	4388.16	3954.42	.00	.00	.00	52379.32
		7	-4303.92	-3585.06	.00	.00	.00	-43563.50
3	6	6	441.76	-412.96	.00	.00	.00	-10382.73
		7	-441.76	412.96	.00	.00	.00	9416.99

42

MEMBER END FORCES      STRUCTURE TYPE = PLANE

-----  
 ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
4		6	977.79	250.70	.00	.00	.00	13058.98
		7	-977.79	-250.70	.00	.00	.00	-12472.70
5		6	2048.58	3671.19	.00	.00	.00	20436.54
		7	-1881.66	-2939.31	.00	.00	.00	-12708.29
6		6	782.82	705.45	.00	.00	.00	9344.20
		7	-767.80	-639.56	.00	.00	.00	-7771.51
7		6	17748.23	18502.47	.00	.00	.00	192771.70
		7	-17134.32	-15810.69	.00	.00	.00	-152651.50
8		6	19167.78	18340.21	.00	.00	.00	195447.90
		7	-18553.87	-15648.43	.00	.00	.00	-155707.30
7	1	7	8383.17	9169.66	.00	.00	.00	85552.38
		8	-7990.76	-7039.42	.00	.00	.00	-57368.63
2	7	7	4149.34	3762.95	.00	.00	.00	43563.39
		8	-4047.28	-3208.91	.00	.00	.00	-31441.11
3	7	7	458.73	-394.06	.00	.00	.00	-9416.97
		8	-458.73	394.06	.00	.00	.00	8046.62
4	7	7	966.42	291.52	.00	.00	.00	12472.70
		8	-46.10	-461.06	.00	.00	.00	-11153.38
5	7	7	1756.37	3016.04	.00	.00	.00	12708.18
		8	-1573.04	-2020.82	.00	.00	.00	-4006.55
6	7	7	740.22	671.29	.00	.00	.00	7771.50
		8	-722.02	-572.45	.00	.00	.00	-5608.93
7	7	7	16454.26	16517.40	.00	.00	.00	152651.20
		8	-14837.92	-12908.60	.00	.00	.00	-101532.00
8	7	7	17879.41	16414.87	.00	.00	.00	155706.90
		8	-15342.75	-12975.60	.00	.00	.00	-104638.70
8	1	8	7811.48	7237.90	.00	.00	.00	57368.85
		9	-7528.00	-5447.76	.00	.00	.00	-35469.99
2	8	8	3965.42	3309.48	.00	.00	.00	31441.05
		9	-3877.94	-2757.06	.00	.00	.00	-20968.56
3	8	8	468.50	-382.44	.00	.00	.00	-8046.66
		9	-468.50	382.44	.00	.00	.00	6726.35
4	8	8	34.46	462.10	.00	.00	.00	11153.38
		9	-34.46	-462.10	.00	.00	.00	-9558.08
5	8	8	1521.79	2059.70	.00	.00	.00	4006.71
		9	-1393.40	-1248.98	.00	.00	.00	1656.35
6	8	8	707.41	590.41	.00	.00	.00	5608.94
		9	-691.81	-491.86	.00	.00	.00	-3740.68
7	8	8	14509.06	13277.14	.00	.00	.00	101532.30
		9	-13994.11	-10025.31	.00	.00	.00	-61354.62
8	8	8	15012.02	13356.80	.00	.00	.00	104639.00
		9	-14497.07	-10104.97	.00	.00	.00	-64186.35
9	1	9	7382.54	5643.35	.00	.00	.00	35470.19
		10	-7183.34	-4129.39	.00	.00	.00	-18614.97

## MEMBER END FORCES      STRUCTURE TYPE = PLANE

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ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
2	9		3804.31	2857.85	.00	.00	.00	20968.55
	10		-3731.41	-2303.81	.00	.00	.00	-12066.03
3	9		478.37	-369.99	.00	.00	.00	-6726.39
	10		-478.37	369.99	.00	.00	.00	5450.08
4	9		22.35	462.82	.00	.00	.00	9558.05
	10		-22.35	-462.82	.00	.00	.00	-7961.60
5	9		1360.18	1285.05	.00	.00	.00	-1656.23
	10		-1274.79	-636.10	.00	.00	.00	4924.11
6	9		678.67	509.82	.00	.00	.00	3740.69
	10		-665.66	-410.98	.00	.00	.00	-2152.52
7	9		13726.42	10388.90	.00	.00	.00	61354.86
	10		-13355.92	-7573.11	.00	.00	.00	-30420.92
8	9		14227.14	10481.73	.00	.00	.00	64186.52
	10		-13856.64	-7665.94	.00	.00	.00	-32932.44
10	1	10	6959.84	4495.73	.00	.00	.00	18615.20
		11	-6858.60	-3217.06	.00	.00	.00	-5423.26
	2	10	3607.08	2493.95	.00	.00	.00	12065.97
		11	-3563.34	-1941.53	.00	.00	.00	-4479.84
	3	10	496.87	-344.73	.00	.00	.00	-5450.12
		11	-496.87	344.73	.00	.00	.00	4270.92
	4	10	-1.64	463.34	.00	.00	.00	7961.56
		11	1.64	-463.34	.00	.00	.00	-6376.62
	5	10	1240.11	701.27	.00	.00	.00	-4924.06
		11	-1200.62	-202.56	.00	.00	.00	6430.58
	6	10	643.49	444.91	.00	.00	.00	2152.52
		11	-635.68	-346.36	.00	.00	.00	-799.16
	7	10	12945.75	8254.48	.00	.00	.00	30421.05
		11	-12753.48	-5826.13	.00	.00	.00	-6377.38
	8	10	13440.99	8373.08	.00	.00	.00	32932.49
		11	-13248.71	-5944.73	.00	.00	.00	-8483.08
11	1	11	6983.67	2935.83	.00	.00	.00	5423.45
		12	-6852.13	-1841.86	.00	.00	.00	2780.95
	2	11	3639.30	1795.14	.00	.00	.00	4479.76
		12	-3572.88	-1242.72	.00	.00	.00	737.15
	3	11	482.47	-364.63	.00	.00	.00	-4270.93
		12	-482.47	364.63	.00	.00	.00	3018.61
	4	11	17.20	463.04	.00	.00	.00	6376.61
		12	-17.20	-463.04	.00	.00	.00	-4786.31
	5	11	1207.87	153.62	.00	.00	.00	-6430.57
		12	-1163.28	217.22	.00	.00	.00	6287.67
	6	11	649.23	320.25	.00	.00	.00	799.18
		12	-637.38	-221.70	.00	.00	.00	131.51
	7	11	12979.74	5303.25	.00	.00	.00	6377.50
		12	-12725.35	-3187.47	.00	.00	.00	8169.59
	8	11	13479.40	5401.66	.00	.00	.00	8483.18
		12	-13225.01	-3285.88	.00	.00	.00	6401.90

40

## MEMBER END FORCES      STRUCTURE TYPE = PLANE

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ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z	
12	1	12	6636.04	2511.32	.00	.00	.00	-2780.86	
		13	-6616.43	-1553.07	.00	.00	.00	9732.53	
	2	12	3432.25	1590.47	.00	.00	.00	-737.11	
		13	-3420.91	-1036.43	.00	.00	.00	5230.05	
	3	12	516.19	-315.05	.00	.00	.00	-3018.61	
		13	-516.19	315.05	.00	.00	.00	1940.89	
	4	12	-28.77	462.46	.00	.00	.00	4786.29	
		13	28.77	-462.46	.00	.00	.00	-3204.33	
	5	12	1179.08	-100.97	.00	.00	.00	-6287.72	
		13	-1173.67	365.16	.00	.00	.00	5462.63	
	6	12	612.30	283.73	.00	.00	.00	-131.50	
		13	-610.27	-184.89	.00	.00	.00	933.01	
	7	12	12347.09	4431.95	.00	.00	.00	-8169.51	
		13	-12308.70	-2556.63	.00	.00	.00	20094.78	
	8	12	12834.51	4579.36	.00	.00	.00	-6401.83	
		13	-12796.13	-2704.04	.00	.00	.00	18831.34	
	13	1	13	6646.97	1416.64	.00	.00	.00	-9732.49
			14	-6611.39	-549.98	.00	.00	.00	13088.34
		2	13	3441.50	965.85	.00	.00	.00	-5230.03
			14	-3418.82	-413.43	.00	.00	.00	7583.61
		3	13	509.61	-325.60	.00	.00	.00	-1940.89
			14	-509.61	325.60	.00	.00	.00	829.65
		4	13	-19.23	462.96	.00	.00	.00	3204.33
			14	19.23	-462.96	.00	.00	.00	-1624.33
5		13	1165.92	-389.20	.00	.00	.00	-5462.70	
		14	-1158.78	563.11	.00	.00	.00	3814.33	
6		13	613.95	172.30	.00	.00	.00	-933.01	
		14	-609.90	-73.75	.00	.00	.00	1352.89	
7		13	12358.71	2302.94	.00	.00	.00	-20094.79	
		14	-12289.27	-611.40	.00	.00	.00	25044.49	
8		13	12849.10	2440.29	.00	.00	.00	-18831.35	
		14	-12779.65	-748.76	.00	.00	.00	24249.81	
14		1	14	6598.68	685.77	.00	.00	.00	-13088.52
			15	-6581.84	137.12	.00	.00	.00	14027.01
		2	14	3409.59	483.68	.00	.00	.00	-7583.62
			15	-3398.25	70.36	.00	.00	.00	8290.46
		3	14	516.21	-315.05	.00	.00	.00	-829.65
			15	-516.21	315.05	.00	.00	.00	-248.06
		4	14	-28.75	462.46	.00	.00	.00	1624.33
			15	28.75	-462.46	.00	.00	.00	-42.38
	5	14	1170.11	-539.15	.00	.00	.00	-3814.33	
		15	-1168.02	641.75	.00	.00	.00	1776.96	
	6	14	608.25	86.28	.00	.00	.00	-1352.88	
		15	-606.23	12.56	.00	.00	.00	1478.97	

45

MEMBER END FORCES      STRUCTURE TYPE = PLANE

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 ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
	7	14	12274.10	863.99	.00	.00	.00	-25044.66
		15	-12241.80	714.38	.00	.00	.00	25282.97
	8	14	12761.56	1011.40	.00	.00	.00	-24249.97
		15	-12729.26	566.97	.00	.00	.00	24992.52
15	1	15	6581.92	132.78	.00	.00	.00	-14027.09
		16	-6598.77	687.70	.00	.00	.00	13080.94
	2	15	3398.28	68.90	.00	.00	.00	-8290.42
		16	-3409.62	483.52	.00	.00	.00	7583.44
	3	15	528.68	-293.64	.00	.00	.00	248.06
		16	-528.68	293.64	.00	.00	.00	-1249.58
	4	15	-47.67	460.90	.00	.00	.00	42.38
		16	47.67	-460.90	.00	.00	.00	1529.61
	5	15	1193.33	-593.35	.00	.00	.00	-1776.96
		16	-1194.38	644.51	.00	.00	.00	-345.66
	6	15	606.24	12.30	.00	.00	.00	-1478.97
		16	-608.26	86.25	.00	.00	.00	1352.85
	7	15	12260.79	-212.11	.00	.00	.00	-25282.99
		16	-12292.04	1734.71	.00	.00	.00	21951.59
	8	15	12741.80	-44.86	.00	.00	.00	-24992.55
		16	-12773.05	1567.45	.00	.00	.00	22231.62
16	1	16	6611.42	-552.99	.00	.00	.00	-13080.87
		17	-6647.00	1422.18	.00	.00	.00	9700.45
	2	16	3418.77	-413.87	.00	.00	.00	-7583.40
		17	-3441.45	967.91	.00	.00	.00	5218.57
	3	16	534.55	-282.79	.00	.00	.00	1249.58
		17	-534.55	282.79	.00	.00	.00	-2217.55
	4	16	-57.08	459.83	.00	.00	.00	-1529.61
		17	57.08	-459.83	.00	.00	.00	3103.53
	5	16	1207.28	-620.02	.00	.00	.00	345.67
		17	-1208.01	632.85	.00	.00	.00	-2500.14
	6	16	609.89	-73.84	.00	.00	.00	-1352.85
		17	-613.94	172.68	.00	.00	.00	930.95
	7	16	12324.83	-1483.69	.00	.00	.00	-21951.48
		17	-12387.87	3018.58	.00	.00	.00	14235.81
	8	16	12802.30	-1306.65	.00	.00	.00	-22231.50
		17	-12865.34	2841.55	.00	.00	.00	15121.79
17	1	17	6616.62	-1557.41	.00	.00	.00	-9700.37
		18	-6636.24	2512.86	.00	.00	.00	2759.31
	2	17	3421.00	-1037.89	.00	.00	.00	-5218.52
		18	-3432.34	1590.31	.00	.00	.00	736.60
	3	17	528.68	-293.63	.00	.00	.00	2217.55
		18	-528.68	293.63	.00	.00	.00	-3219.07
	4	17	-47.67	460.89	.00	.00	.00	-3103.54
		18	47.67	-460.89	.00	.00	.00	4675.51

46

MEMBER END FORCES      STRUCTURE TYPE = PLANE

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 ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z	
5	17	18	1194.85	-657.34	.00	.00	.00	2500.15	
		18	-1194.83	654.78	.00	.00	.00	-4736.33	
	6	17	610.29	-185.17	.00	.00	.00	-930.97	
		18	-612.31	283.71	.00	.00	.00	131.38	
	7	17	12323.78	-3270.54	.00	.00	.00	-14235.70	
		18	-12356.73	4874.40	.00	.00	.00	347.40	
8	17	12804.79	-3103.28	.00	.00	.00	-15121.69		
	18	-12837.74	4707.14	.00	.00	.00	1803.85		
18	1	18	6852.36	-1843.75	.00	.00	.00	-2759.35	
		19	-6983.90	2937.72	.00	.00	.00	-5451.88	
	2	18	3572.89	-1242.75	.00	.00	.00	-736.53	
		19	-3639.31	1795.17	.00	.00	.00	-4480.60	
	3	18	555.15	-239.87	.00	.00	.00	3219.06	
		19	-555.15	239.87	.00	.00	.00	-4042.90	
	4	18	-93.07	453.91	.00	.00	.00	-4675.50	
		19	93.07	-453.91	.00	.00	.00	6234.51	
	5	18	1253.77	-533.32	.00	.00	.00	4736.34	
		19	-1254.70	540.99	.00	.00	.00	-6576.82	
	6	18	637.38	-221.70	.00	.00	.00	-131.39	
		19	-649.23	320.25	.00	.00	.00	-799.31	
	7	18	12778.49	-3627.46	.00	.00	.00	-347.37	
		19	-12989.21	5380.07	.00	.00	.00	-15116.99	
	8	18	13240.56	-3413.42	.00	.00	.00	-1803.81	
		19	-13451.29	5166.03	.00	.00	.00	-12925.38	
	19	1	19	6858.00	-3220.52	.00	.00	.00	5451.88
			20	-6959.25	4502.93	.00	.00	.00	-18700.35
		2	19	3562.91	-1942.53	.00	.00	.00	4480.46
			20	-3606.65	2496.57	.00	.00	.00	-12094.77
		3	19	544.88	-262.34	.00	.00	.00	4042.93
			20	-544.88	262.34	.00	.00	.00	-4942.95
		4	19	-74.44	457.35	.00	.00	.00	-6234.49
			20	74.44	-457.35	.00	.00	.00	7803.51
5		19	1231.54	-591.81	.00	.00	.00	6576.83	
		20	-1233.97	622.59	.00	.00	.00	-8651.11	
6		19	635.60	-346.52	.00	.00	.00	799.30	
		20	-643.41	445.36	.00	.00	.00	-2157.64	
7		19	12758.49	-5906.37	.00	.00	.00	15116.91	
		20	-12913.70	7872.44	.00	.00	.00	-38743.31	
8		19	13228.92	-5711.35	.00	.00	.00	12925.35	
		20	-13384.14	7677.43	.00	.00	.00	-35882.76	
20		1	20	7185.65	-4132.15	.00	.00	.00	18700.46
			21	-7384.85	5641.69	.00	.00	.00	-35509.62
		2	20	3732.50	-2304.05	.00	.00	.00	12094.83
			21	-3805.40	2856.46	.00	.00	.00	-20969.63

47

MEMBER END FORCES      STRUCTURE TYPE = PLANE

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ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
3		20	557.88	-233.44	.00	.00	.00	4942.96
		21	-557.88	233.44	.00	.00	.00	-5745.90
4		20	-98.32	452.83	.00	.00	.00	-7803.44
		21	98.32	-452.83	.00	.00	.00	9360.95
5		20	1264.89	-557.07	.00	.00	.00	8651.10
		21	-1274.34	628.68	.00	.00	.00	-10675.72
6		20	665.86	-411.03	.00	.00	.00	2157.65
		21	-678.86	509.58	.00	.00	.00	-3740.89
7		20	13308.45	-7184.91	.00	.00	.00	38743.57
		21	-13603.02	9417.03	.00	.00	.00	-67280.80
8		20	13768.01	-6965.52	.00	.00	.00	35883.10
		21	-14062.57	9197.63	.00	.00	.00	-63665.74
21	1	21	7525.70	-5452.30	.00	.00	.00	35509.72
		22	-7809.18	7247.69	.00	.00	.00	-57495.72
2	21	21	3876.72	-2758.93	.00	.00	.00	20969.57
		22	-3964.20	3312.97	.00	.00	.00	-31481.39
3	21	21	563.64	-219.17	.00	.00	.00	5745.95
		22	-563.64	219.17	.00	.00	.00	-6504.82
4	21	21	-109.76	450.15	.00	.00	.00	-9360.94
		22	109.76	-450.15	.00	.00	.00	10919.51
5	21	21	1289.90	-596.14	.00	.00	.00	10675.71
		22	-1310.55	726.95	.00	.00	.00	-12946.90
6	21	21	691.59	-492.17	.00	.00	.00	3740.89
		22	-707.19	591.01	.00	.00	.00	-5616.09
7	21	21	13837.79	-9068.56	.00	.00	.00	67280.91
		22	-14245.02	11647.64	.00	.00	.00	-103125.40
8	21	21	14291.68	-8837.58	.00	.00	.00	63665.91
		22	-14698.91	11416.66	.00	.00	.00	-98710.71
22	1	22	7995.60	-7041.58	.00	.00	.00	57495.90
		23	-8388.02	9165.60	.00	.00	.00	-85596.87
2	22	22	4049.27	-3208.44	.00	.00	.00	31481.24
		23	-4151.33	3760.86	.00	.00	.00	-43565.05
3	22	22	569.15	-204.42	.00	.00	.00	6504.84
		23	-569.15	204.42	.00	.00	.00	-7213.78
4	22	22	-121.45	447.14	.00	.00	.00	-10919.64
		23	121.45	-447.14	.00	.00	.00	12470.12
5	22	22	1329.06	-692.53	.00	.00	.00	12946.87
		23	-1367.33	899.68	.00	.00	.00	-15682.36
6	22	22	722.37	-572.38	.00	.00	.00	5616.10
		23	-740.58	670.93	.00	.00	.00	-7771.78
7	22	22	14544.00	-11272.21	.00	.00	.00	103125.30
		23	-15094.95	14254.34	.00	.00	.00	-147359.70
8	22	22	14991.70	-11029.48	.00	.00	.00	98710.52
		23	-15542.66	14011.62	.00	.00	.00	-142103.40
23	1	23	8753.11	-8817.52	.00	.00	.00	85596.76
		24	-9100.84	10348.86	.00	.00	.00	-108101.80

68

## MEMBER END FORCES      STRUCTURE TYPE = PLANE

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ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
	2	23	4300.53	-3589.18	.00	.00	.00	43564.98
		24	-4384.77	3960.16	.00	.00	.00	-52429.24
	3	23	576.98	-181.06	.00	.00	.00	7213.70
		24	-576.98	181.06	.00	.00	.00	-7639.16
	4	23	-139.51	441.85	.00	.00	.00	-12470.34
		24	139.51	-441.85	.00	.00	.00	13507.72
	5	23	1402.74	-843.36	.00	.00	.00	15682.47
		24	-1441.35	1013.39	.00	.00	.00	-17861.89
	6	23	767.20	-640.31	.00	.00	.00	7771.77
		24	-782.22	706.49	.00	.00	.00	-9353.11
	7	23	15661.06	-13629.59	.00	.00	.00	147359.30
		24	-16146.65	15768.12	.00	.00	.00	-181877.50
	8	23	16098.53	-13368.80	.00	.00	.00	142102.70
		24	-16584.13	15507.33	.00	.00	.00	-176008.90
24	1	24	11622.73	-7405.53	.00	.00	.00	-108101.40
		25	-12750.77	7483.77	.00	.00	.00	95191.47
	2	24	4586.10	-3725.11	.00	.00	.00	-52429.08
		25	-4586.10	3725.11	.00	.00	.00	45969.27
	3	24	267.54	-541.65	.00	.00	.00	-7638.30
		25	-267.54	541.65	.00	.00	.00	6699.51
	4	24	-458.02	70.45	.00	.00	.00	13509.32
		25	454.46	-121.83	.00	.00	.00	-13294.56
	5	24	1222.60	-1269.66	.00	.00	.00	-17862.68
		25	-1222.60	1269.66	.00	.00	.00	15661.28
	6	24	818.14	-664.54	.00	.00	.00	-9353.08
		25	-818.14	664.54	.00	.00	.00	8200.70
	7	24	18059.09	-13536.04	.00	.00	.00	-181875.20
		25	-19190.69	13562.90	.00	.00	.00	158427.70
	8	24	17868.61	-14007.24	.00	.00	.00	-176004.20
		25	-19003.77	13982.72	.00	.00	.00	151832.60
25	1	25	12770.84	-7449.23	.00	.00	.00	-95191.50
		26	-14608.22	7571.72	.00	.00	.00	72610.01
	2	25	4596.09	-3712.92	.00	.00	.00	-45969.09
		26	-4596.09	3712.92	.00	.00	.00	34805.53
	3	25	268.99	-541.69	.00	.00	.00	-6698.69
		26	-268.99	541.69	.00	.00	.00	5070.20
	4	25	-454.79	120.37	.00	.00	.00	13296.35
		26	424.73	-571.27	.00	.00	.00	-12211.97
	5	25	1226.00	-1265.39	.00	.00	.00	-15661.42
		26	-1226.00	1265.39	.00	.00	.00	11856.92
	6	25	819.92	-662.36	.00	.00	.00	-8200.64
		26	-819.92	662.36	.00	.00	.00	6209.15
	7	25	19227.06	-13511.22	.00	.00	.00	-158425.00
		26	-21094.50	13182.81	.00	.00	.00	118339.80
	8	25	19041.27	-13932.54	.00	.00	.00	-151827.30
		26	-20938.77	13153.23	.00	.00	.00	111198.10



49

MEMBER END FORCES      STRUCTURE TYPE = PLANE

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 ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
26	1	26	14583.01	-7620.25	.00	.00	.00	-72610.19
		27	-16236.03	7735.96	.00	.00	.00	49519.45
	2	26	4583.74	-3728.12	.00	.00	.00	-34805.52
		27	-4583.74	3728.12	.00	.00	.00	23593.74
	3	26	267.19	-542.53	.00	.00	.00	-5070.23
		27	-267.19	542.53	.00	.00	.00	3438.52
	4	26	-422.83	572.59	.00	.00	.00	12212.12
		27	378.79	-1201.69	.00	.00	.00	-9499.67
	5	26	1221.80	-1269.58	.00	.00	.00	-11856.70
		27	-1221.80	1269.58	.00	.00	.00	8038.87
	6	26	817.72	-665.09	.00	.00	.00	-6209.14
		27	-817.72	665.09	.00	.00	.00	4208.99
7	26	21050.63	-13252.97	.00	.00	.00	-118339.70	
	27	-22747.69	12739.58	.00	.00	.00	79299.91	
8	26	20895.00	-13222.91	.00	.00	.00	-111197.80	
	27	-22636.09	12080.42	.00	.00	.00	73238.77	
27	1	27	16261.60	-7682.00	.00	.00	.00	-49519.50
		28	-17733.38	7780.12	.00	.00	.00	26274.93
	2	27	4596.08	-3712.93	.00	.00	.00	-23593.72
		28	-4596.08	3712.93	.00	.00	.00	12430.17
	3	27	268.99	-541.64	.00	.00	.00	-3438.55
		28	-268.99	541.64	.00	.00	.00	1810.09
	4	27	-382.78	1200.36	.00	.00	.00	9499.67
		28	382.78	-1200.36	.00	.00	.00	-5890.61
	5	27	1226.00	-1265.48	.00	.00	.00	-8038.85
		28	-1226.00	1265.48	.00	.00	.00	4233.86
	6	27	819.92	-662.37	.00	.00	.00	-4209.00
		28	-819.92	662.37	.00	.00	.00	2217.48
7	27	22789.82	-12664.06	.00	.00	.00	-79299.95	
	28	-24261.60	12762.18	.00	.00	.00	41075.91	
8	27	22676.03	-12005.33	.00	.00	.00	-73238.83	
	28	-24147.81	12103.46	.00	.00	.00	36995.39	
28	1	28	17714.73	-7822.50	.00	.00	.00	-26274.92
		29	-19136.83	7920.73	.00	.00	.00	-.03
	2	28	4587.19	-3723.92	.00	.00	.00	-12430.18
		29	-4587.19	3723.92	.00	.00	.00	-.01
	3	28	267.70	-542.28	.00	.00	.00	-1810.07
		29	-267.70	542.28	.00	.00	.00	.01
	4	28	-379.90	1201.28	.00	.00	.00	5890.58
		29	304.35	-2295.19	.00	.00	.00	.01
	5	28	1222.97	-1268.42	.00	.00	.00	-4233.88
		29	-1222.97	1268.42	.00	.00	.00	.02
	6	28	818.33	-664.33	.00	.00	.00	-2217.48
		29	-818.33	664.33	.00	.00	.00	.00

50

MEMBER END FORCES      STRUCTURE TYPE = PLANE

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ALL UNITS ARE -- POUN FEET

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
	7	28	24231.02	-12820.17	.00	.00	.00	-41075.96
		29	-25728.67	11824.49	.00	.00	.00	.00
	8	28	24118.81	-12161.17	.00	.00	.00	-36995.45
		29	-25692.02	10071.58	.00	.00	.00	.02

\*\*\*\*\* END OF LATEST ANALYSIS RESULT \*\*\*\*\*

\*\*\* STAAD-III ERROR MESSAGE \*\*\*  
NO END OR FINISH STATEMENT FOUND. RUN TERMINATED.

\*\*\*\*\* END OF STAAD-III \*\*\*\*\*

\*\*\*\* DATE= SEP 2,1998    TIME= 11: 3:52 \*\*\*\*

\*\*\*\*\*  
\*            For questions on STAAD-III,    contact:            \*  
\*                      Research Engineers, Inc at            \*  
\* West Coast: Ph- (714) 974-2500 Fax- (714) 921-2543 \*  
\* East Coast: Ph- (508) 688-3626 Fax- (508) 685-7230 \*  
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*****
*
*           S T A A D - III
*           Revision 22.3a
*           Proprietary Program of
*           Research Engineers, Inc.
*           Date=    AUG 28, 1998
*           Time=    8:23:29
*
*           USER ID: Rust E & I
*****

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1. STAAD PLANE CEMETERY ROAD UNDERPASS
2. INPUT WIDTH 72
3. UNIT FEET KIP
4. JOINT COORDINATES
5. 1 0.0 0.0; 2 -0.23 3.33; 3 -0.43 6.33; 4 -0.63 9.33
6. 5 -0.83 12.33; 6 -0.95 14.06; 7 1.33 14.58; 8 4.75 15.21
7. 9 8.16 15.75; 10 11.58 16.20; 11 14.99 16.47; 12 18.40 16.88
8. 13 21.82 17.09; 14 25.23 17.09; 15 28.65 17.16; 16 32.06 17.09
9. 17 35.48 16.95; 18 38.89 16.88; 19 42.30 16.47; 20 45.72 16.20
10. 21 49.13 15.75; 22 52.55 15.21; 23 55.96 14.58; 24 58.25 14.06
11. 25 58.13 12.33; 26 57.93 9.33; 27 57.72 6.33; 28 57.52 3.33
12. 29 57.29 0.0
13. MEMBER INCIDENCE
14. 1 1 2; 2 2 3; 3 3 4; 4 4 5; 5 5 6; 6 6 7; 7 7 8; 8 8 9
15. 9 9 10; 10 10 11; 11 11 12; 12 12 13; 13 13 14; 14 14 15
16. 15 15 16; 16 16 17; 17 17 18; 18 18 19; 19 19 20; 20 20 21
17. 21 21 22; 22 22 23; 23 23 24; 24 24 25; 25 25 26; 26 26 27
18. 27 27 28; 28 28 29
19. UNIT INCHES KIP
20. MEMBER PROPERTY AMERICAN
21. 1 28 PRI AX 410. IX 281.52 IZ 1126.06
22. 2 27 PRI AX 471. IX 424.3 IZ 1697.19
23. 3 26 PRI AX 529. IX 601.16 IZ 2404.63
24. 4 25 PRI AX 588. IX 820.88 IZ 3283.53
25. 5 24 PRI AX 626. IX 988.52 IZ 3954.09
26. 6 23 PRI AX 642. IX 1064.86 IZ 4259.42
27. 7 22 PRI AX 598. IX 875.98 IZ 3503.93
28. 8 21 PRI AX 504. IX 525.84 IZ 2103.36
29. 9 20 PRI AX 425. IX 314.71 IZ 1258.82
30. 10 19 PRI AX 360. IX 191.1 IZ 764.4
31. 11 18 PRI AX 308. IX 119.58 IZ 478.31
32. 12 17 PRI AX 269. IX 79.29 IZ 317.14
33. 13 16 PRI AX 244. IX 58.7 IZ 234.78
34. 14 15 PRI AX 231. IX 49.65 IZ 198.6
35. CONSTANT
36. E CONCRETE ALL
37. UNIT FEET POUND
38. CONSTANT
39. DENSITY CONCRETE ALL
40. UNIT FEET KIP
41. SUPPORT
42. 1 29 PINNED
43. UNIT FEET POUND
44. LOAD 1
45. JOINT LOAD
46. 7 FY -1
47. LOAD 2
48. JOINT LOAD

52

- 49. 8 FY -1
- 50. LOAD 3
- 51. JOINT LOAD
- 52. 9 FY -1
- 53. LOAD 4
- 54. JOINT LOAD
- 55. 10 FY -1
- 56. LOAD 5
- 57. JOINT LOAD
- 58. 11 FY -1
- 59. LOAD 6
- 60. JOINT LOAD
- 61. 12 FY -1
- 62. LOAD 7
- 63. JOINT LOAD
- 64. 13 FY -1
- 65. LOAD 8
- 66. JOINT LOAD
- 67. 14 FY -1
- 68. LOAD 9
- 69. JOINT LOAD
- 70. 15 FY -1
- 71. PERFORM ANALYSIS

P R O B L E M   S T A T I S T I C S

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NUMBER OF JOINTS/MEMBER+ELEMENTS/SUPPORTS =    29/    28/    2  
ORIGINAL/FINAL BAND-WIDTH =    1/    1  
TOTAL PRIMARY LOAD CASES =    9, TOTAL DEGREES OF FREEDOM =    83  
SIZE OF STIFFNESS MATRIX =    498 DOUBLE PREC. WORDS  
REQRD/AVAIL. DISK SPACE = 12.05/ 388.3 MB, EXMEM = 1964.9 MB

++ Processing Element Stiffness Matrix.                    8:23:30  
++ Processing Global Stiffness Matrix.                    8:23:30  
++ Processing Triangular Factorization.                    8:23:30  
++ Calculating Joint Displacements.                        8:23:30  
++ Calculating Member Forces.                              8:23:31

72. PRINT MEMBER FORCES ALL

## MEMBER END FORCES

STRUCTURE TYPE = PLANE

53

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
1	1	1	.97	.11	.00	.00	.00	.00
		2	-.97	-.11	.00	.00	.00	.35
	2	1	.91	.20	.00	.00	.00	.00
		2	-.91	-.20	.00	.00	.00	.66
	3	1	.84	.29	.00	.00	.00	.00
		2	-.84	-.29	.00	.00	.00	.97
	4	1	.77	.38	.00	.00	.00	.00
		2	-.77	-.38	.00	.00	.00	1.27
	5	1	.71	.46	.00	.00	.00	.00
		2	-.71	-.46	.00	.00	.00	1.54
	6	1	.64	.54	.00	.00	.00	.00
		2	-.64	-.54	.00	.00	.00	1.79
	7	1	.58	.59	.00	.00	.00	.00
		2	-.58	-.59	.00	.00	.00	1.98
	8	1	.52	.63	.00	.00	.00	.00
		2	-.52	-.63	.00	.00	.00	2.10
	9	1	.46	.64	.00	.00	.00	.00
		2	-.46	-.64	.00	.00	.00	2.14
2	1	2	.97	.10	.00	.00	.00	-.35
		3	-.97	-.10	.00	.00	.00	.66
	2	2	.91	.20	.00	.00	.00	-.66
		3	-.91	-.20	.00	.00	.00	1.26
	3	2	.84	.29	.00	.00	.00	-.97
		3	-.84	-.29	.00	.00	.00	1.84
	4	2	.77	.38	.00	.00	.00	-1.27
		3	-.77	-.38	.00	.00	.00	2.40
	5	2	.71	.46	.00	.00	.00	-1.54
		3	-.71	-.46	.00	.00	.00	2.93
	6	2	.64	.53	.00	.00	.00	-1.79
		3	-.64	-.53	.00	.00	.00	3.39
	7	2	.58	.59	.00	.00	.00	-1.98
		3	-.58	-.59	.00	.00	.00	3.76
	8	2	.52	.63	.00	.00	.00	-2.10
		3	-.52	-.63	.00	.00	.00	4.00
	9	2	.46	.64	.00	.00	.00	-2.14
		3	-.46	-.64	.00	.00	.00	4.06
3	1	3	.97	.10	.00	.00	.00	-.66
		4	-.97	-.10	.00	.00	.00	.97
	2	3	.91	.20	.00	.00	.00	-1.26
		4	-.91	-.20	.00	.00	.00	1.85
	3	3	.84	.29	.00	.00	.00	-1.84
		4	-.84	-.29	.00	.00	.00	2.71
	4	3	.77	.38	.00	.00	.00	-2.40
		4	-.77	-.38	.00	.00	.00	3.54
	5	3	.71	.46	.00	.00	.00	-2.93
		4	-.71	-.46	.00	.00	.00	4.31
	6	3	.64	.53	.00	.00	.00	-3.39
		4	-.64	-.53	.00	.00	.00	4.99
	7	3	.58	.59	.00	.00	.00	-3.76
		4	-.58	-.59	.00	.00	.00	5.54
	8	3	.52	.63	.00	.00	.00	-4.00
		4	-.52	-.63	.00	.00	.00	5.89
	9	3	.46	.64	.00	.00	.00	-4.06
		4	-.46	-.64	.00	.00	.00	5.99

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
4	1	4	.97	.10	.00	.00	.00	-.97
		5	-.97	-.10	.00	.00	.00	1.28
	2	4	.91	.20	.00	.00	.00	-1.85
		5	-.91	-.20	.00	.00	.00	2.44
	3	4	.84	.29	.00	.00	.00	-2.71
		5	-.84	-.29	.00	.00	.00	3.58
	4	4	.77	.38	.00	.00	.00	-3.54
		5	-.77	-.38	.00	.00	.00	4.67
	5	4	.71	.46	.00	.00	.00	-4.31
		5	-.71	-.46	.00	.00	.00	5.69
	6	4	.64	.53	.00	.00	.00	-4.99
		5	-.64	-.53	.00	.00	.00	6.60
	7	4	.58	.59	.00	.00	.00	-5.54
		5	-.58	-.59	.00	.00	.00	7.32
	8	4	.52	.63	.00	.00	.00	-5.89
		5	-.52	-.63	.00	.00	.00	7.78
	9	4	.46	.64	.00	.00	.00	-5.99
		5	-.46	-.64	.00	.00	.00	7.91
5	1	5	.97	.11	.00	.00	.00	-1.28
		6	-.97	-.11	.00	.00	.00	1.46
	2	5	.91	.20	.00	.00	.00	-2.44
		6	-.91	-.20	.00	.00	.00	2.79
	3	5	.84	.29	.00	.00	.00	-3.58
		6	-.84	-.29	.00	.00	.00	4.08
	4	5	.77	.38	.00	.00	.00	-4.67
		6	-.77	-.38	.00	.00	.00	5.33
	5	5	.71	.46	.00	.00	.00	-5.69
		6	-.71	-.46	.00	.00	.00	6.50
	6	5	.64	.54	.00	.00	.00	-6.60
		6	-.64	-.54	.00	.00	.00	7.53
	7	5	.58	.59	.00	.00	.00	-7.32
		6	-.58	-.59	.00	.00	.00	8.35
	8	5	.52	.63	.00	.00	.00	-7.78
		6	-.52	-.63	.00	.00	.00	8.87
	9	5	.46	.64	.00	.00	.00	-7.91
		6	-.46	-.64	.00	.00	.00	9.02
6	1	6	.25	.94	.00	.00	.00	1.46
		7	-.25	-.94	.00	.00	.00	.75
	2	6	.34	.86	.00	.00	.00	2.79
		7	-.34	-.86	.00	.00	.00	-.77
	3	6	.42	.78	.00	.00	.00	4.08
		7	-.42	-.78	.00	.00	.00	-2.25
	4	6	.50	.71	.00	.00	.00	5.33
		7	-.50	-.71	.00	.00	.00	-3.68
	5	6	.57	.63	.00	.00	.00	6.50
		7	-.57	-.63	.00	.00	.00	-5.03
	6	6	.63	.55	.00	.00	.00	7.53
		7	-.63	-.55	.00	.00	.00	-6.23
	7	6	.68	.48	.00	.00	.00	8.35
		7	-.68	-.48	.00	.00	.00	-7.23
	8	6	.70	.41	.00	.00	.00	8.87
		7	-.70	-.41	.00	.00	.00	-7.91
	9	6	.70	.35	.00	.00	.00	9.02
		7	-.70	-.35	.00	.00	.00	-8.20

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
7	1	7	.03	-.03	.00	.00	.00	-.75
		8	-.03	.03	.00	.00	.00	.64
	2	7	.30	.88	.00	.00	.00	.77
		8	-.30	-.88	.00	.00	.00	2.28
	3	7	.38	.80	.00	.00	.00	2.25
		8	-.38	-.80	.00	.00	.00	.54
	4	7	.46	.73	.00	.00	.00	3.68
		8	-.46	-.73	.00	.00	.00	-1.16
	5	7	.54	.65	.00	.00	.00	5.03
		8	-.54	-.65	.00	.00	.00	-2.76
	6	7	.60	.58	.00	.00	.00	6.23
		8	-.60	-.58	.00	.00	.00	-4.22
	7	7	.65	.51	.00	.00	.00	7.23
		8	-.65	-.51	.00	.00	.00	-5.46
	8	7	.68	.44	.00	.00	.00	7.91
		8	-.68	-.44	.00	.00	.00	-6.37
9	7	.69	.38	.00	.00	.00	8.20	
	8	-.69	-.38	.00	.00	.00	-6.87	
8	1	8	.03	-.03	.00	.00	.00	-.64
		9	-.03	.03	.00	.00	.00	.54
	2	8	.12	-.10	.00	.00	.00	-2.28
		9	-.12	.10	.00	.00	.00	1.93
	3	8	.36	.81	.00	.00	.00	-.54
		9	-.36	-.81	.00	.00	.00	3.34
	4	8	.45	.74	.00	.00	.00	1.16
		9	-.45	-.74	.00	.00	.00	1.39
	5	8	.52	.66	.00	.00	.00	2.76
		9	-.52	-.66	.00	.00	.00	-.47
	6	8	.59	.59	.00	.00	.00	4.22
		9	-.59	-.59	.00	.00	.00	-2.17
	7	8	.64	.53	.00	.00	.00	5.46
		9	-.64	-.53	.00	.00	.00	-3.64
	8	8	.67	.46	.00	.00	.00	6.37
		9	-.67	-.46	.00	.00	.00	-4.78
	9	8	.68	.40	.00	.00	.00	6.87
		9	-.68	-.40	.00	.00	.00	-5.49
9	1	9	.03	-.03	.00	.00	.00	-.54
		10	-.03	.03	.00	.00	.00	.45
	2	9	.12	-.10	.00	.00	.00	-1.93
		10	-.12	.10	.00	.00	.00	1.58
	3	9	.21	-.17	.00	.00	.00	-3.34
		10	-.21	.17	.00	.00	.00	2.75
	4	9	.43	.75	.00	.00	.00	-1.39
		10	-.43	-.75	.00	.00	.00	3.97
	5	9	.51	.68	.00	.00	.00	.47
		10	-.51	-.68	.00	.00	.00	1.87
	6	9	.57	.61	.00	.00	.00	2.17
		10	-.57	-.61	.00	.00	.00	-.07
	7	9	.63	.54	.00	.00	.00	3.64
		10	-.63	-.54	.00	.00	.00	-1.78
	8	9	.66	.48	.00	.00	.00	4.78
		10	-.66	-.48	.00	.00	.00	-3.13
	9	9	.67	.42	.00	.00	.00	5.49
		10	-.67	-.42	.00	.00	.00	-4.06

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
10	1	10	.04	-.03	.00	.00	.00	-.45
		11	-.04	.03	.00	.00	.00	.36
	2	10	.13	-.09	.00	.00	.00	-1.58
		11	-.13	.09	.00	.00	.00	1.26
	3	10	.22	-.16	.00	.00	.00	-2.75
		11	-.22	.16	.00	.00	.00	2.20
	4	10	.31	-.23	.00	.00	.00	-3.97
		11	-.31	.23	.00	.00	.00	3.19
	5	10	.47	-.70	.00	.00	.00	-1.87
		11	-.47	.70	.00	.00	.00	4.28
	6	10	.54	-.64	.00	.00	.00	.07
		11	-.54	.64	.00	.00	.00	2.11
	7	10	.60	-.57	.00	.00	.00	1.78
		11	-.60	.57	.00	.00	.00	.19
	8	10	.64	-.51	.00	.00	.00	3.13
		11	-.64	.51	.00	.00	.00	-1.38
	9	10	.65	-.45	.00	.00	.00	4.06
		11	-.65	.45	.00	.00	.00	-2.52
11	1	11	.03	-.03	.00	.00	.00	-.36
		12	-.03	.03	.00	.00	.00	.26
	2	11	.13	-.10	.00	.00	.00	-1.26
		12	-.13	.10	.00	.00	.00	.92
	3	11	.21	-.17	.00	.00	.00	-2.20
		12	-.21	.17	.00	.00	.00	1.62
	4	11	.30	-.24	.00	.00	.00	-3.19
		12	-.30	.24	.00	.00	.00	2.37
	5	11	.38	-.31	.00	.00	.00	-4.28
		12	-.38	.31	.00	.00	.00	3.22
	6	11	.57	-.62	.00	.00	.00	-2.11
		12	-.57	.62	.00	.00	.00	4.23
	7	11	.62	-.55	.00	.00	.00	-.19
		12	-.62	.55	.00	.00	.00	2.07
	8	11	.66	-.48	.00	.00	.00	1.38
		12	-.66	.48	.00	.00	.00	.28
	9	11	.66	-.42	.00	.00	.00	2.52
		12	-.66	.42	.00	.00	.00	-1.06
12	1	12	.04	-.03	.00	.00	.00	-.26
		13	-.04	.03	.00	.00	.00	.17
	2	12	.13	-.09	.00	.00	.00	-.92
		13	-.13	.09	.00	.00	.00	.61
	3	12	.22	-.16	.00	.00	.00	-1.62
		13	-.22	.16	.00	.00	.00	1.08
	4	12	.31	-.22	.00	.00	.00	-2.37
		13	-.31	.22	.00	.00	.00	1.61
	5	12	.40	-.29	.00	.00	.00	-3.22
		13	-.40	.29	.00	.00	.00	2.24
	6	12	.47	-.35	.00	.00	.00	-4.23
		13	-.47	.35	.00	.00	.00	3.03
	7	12	.59	-.58	.00	.00	.00	-2.07
		13	-.59	.58	.00	.00	.00	4.07
	8	12	.63	-.52	.00	.00	.00	-.28
		13	-.63	.52	.00	.00	.00	2.07
	9	12	.64	-.46	.00	.00	.00	1.06
		13	-.64	.46	.00	.00	.00	.52



MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
13	1	13	.04	-.02	.00	.00	.00	-.17
		14	-.04	.02	.00	.00	.00	.10
	2	13	.14	-.08	.00	.00	.00	-.61
		14	-.14	.08	.00	.00	.00	.33
	3	13	.23	-.14	.00	.00	.00	-1.08
		14	-.23	.14	.00	.00	.00	.60
	4	13	.33	-.20	.00	.00	.00	-1.61
		14	-.33	.20	.00	.00	.00	.92
	5	13	.41	-.26	.00	.00	.00	-2.24
		14	-.41	.26	.00	.00	.00	1.35
	6	13	.49	-.32	.00	.00	.00	-3.03
		14	-.49	.32	.00	.00	.00	1.93
	7	13	.55	-.38	.00	.00	.00	-4.07
		14	-.55	.38	.00	.00	.00	2.77
	8	13	.59	-.56	.00	.00	.00	-2.07
		14	-.59	.56	.00	.00	.00	3.98
	9	13	.61	-.50	.00	.00	.00	-.52
		14	-.61	.50	.00	.00	.00	2.22
14	1	14	.04	-.02	.00	.00	.00	-.10
		15	-.04	.02	.00	.00	.00	.01
	2	14	.13	-.09	.00	.00	.00	-.33
		15	-.13	.09	.00	.00	.00	.04
	3	14	.23	-.15	.00	.00	.00	-.60
		15	-.23	.15	.00	.00	.00	.09
	4	14	.32	-.21	.00	.00	.00	-.92
		15	-.32	.21	.00	.00	.00	.21
	5	14	.41	-.27	.00	.00	.00	-1.35
		15	-.41	.27	.00	.00	.00	.42
	6	14	.48	-.33	.00	.00	.00	-1.93
		15	-.48	.33	.00	.00	.00	.80
	7	14	.54	-.39	.00	.00	.00	-2.77
		15	-.54	.39	.00	.00	.00	1.43
	8	14	.58	-.45	.00	.00	.00	-3.98
		15	-.58	.45	.00	.00	.00	2.43
	9	14	.62	-.49	.00	.00	.00	-2.22
		15	-.62	.49	.00	.00	.00	3.89
15	1	15	.04	-.02	.00	.00	.00	-.01
		16	-.04	.02	.00	.00	.00	-.06
	2	15	.14	-.08	.00	.00	.00	-.04
		16	-.14	.08	.00	.00	.00	-.24
	3	15	.24	-.14	.00	.00	.00	-.09
		16	-.24	.14	.00	.00	.00	-.38
	4	15	.33	-.20	.00	.00	.00	-.21
		16	-.33	.20	.00	.00	.00	-.46
	5	15	.42	-.25	.00	.00	.00	-.42
		16	-.42	.25	.00	.00	.00	-.44
	6	15	.50	-.31	.00	.00	.00	-.80
		16	-.50	.31	.00	.00	.00	-.26
	7	15	.56	-.37	.00	.00	.00	-1.43
		16	-.56	.37	.00	.00	.00	.17
	8	15	.60	-.43	.00	.00	.00	-2.43
		16	-.60	.43	.00	.00	.00	.97
	9	15	.62	-.49	.00	.00	.00	-3.89
		16	-.62	.49	.00	.00	.00	2.23

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
16	1	16	.04	-.02	.00	.00	.00	.06
		17	-.04	.02	.00	.00	.00	-.14
	2	16	.14	-.08	.00	.00	.00	.24
		17	-.14	.08	.00	.00	.00	-.50
	3	16	.24	-.13	.00	.00	.00	.38
		17	-.24	.13	.00	.00	.00	-.83
	4	16	.33	-.19	.00	.00	.00	.46
		17	-.33	.19	.00	.00	.00	-1.10
	5	16	.42	-.24	.00	.00	.00	.44
		17	-.42	.24	.00	.00	.00	-1.28
	6	16	.50	-.30	.00	.00	.00	.26
		17	-.50	.30	.00	.00	.00	-1.29
	7	16	.57	-.36	.00	.00	.00	-.17
		17	-.57	.36	.00	.00	.00	-1.05
	8	16	.61	-.42	.00	.00	.00	-.97
		17	-.61	.42	.00	.00	.00	-.45
	9	16	.63	-.47	.00	.00	.00	-2.23
		17	-.63	.47	.00	.00	.00	.60
17	1	17	.04	-.02	.00	.00	.00	.14
		18	-.04	.02	.00	.00	.00	-.21
	2	17	.14	-.08	.00	.00	.00	.50
		18	-.14	.08	.00	.00	.00	-.77
	3	17	.24	-.14	.00	.00	.00	.83
		18	-.24	.14	.00	.00	.00	-1.30
	4	17	.33	-.20	.00	.00	.00	1.10
		18	-.33	.20	.00	.00	.00	-1.77
	5	17	.42	-.25	.00	.00	.00	1.28
		18	-.42	.25	.00	.00	.00	-2.14
	6	17	.50	-.31	.00	.00	.00	1.29
		18	-.50	.31	.00	.00	.00	-2.35
	7	17	.56	-.37	.00	.00	.00	1.05
		18	-.56	.37	.00	.00	.00	-2.31
	8	17	.60	-.43	.00	.00	.00	.45
		18	-.60	.43	.00	.00	.00	-1.91
	9	17	.62	-.49	.00	.00	.00	-.60
		18	-.62	.49	.00	.00	.00	-1.06
18	1	18	.04	-.02	.00	.00	.00	.21
		19	-.04	.02	.00	.00	.00	-.28
	2	18	.15	-.07	.00	.00	.00	.77
		19	-.15	.07	.00	.00	.00	-1.00
	3	18	.25	-.11	.00	.00	.00	1.30
		19	-.25	.11	.00	.00	.00	-1.69
	4	18	.35	-.16	.00	.00	.00	1.77
		19	-.35	.16	.00	.00	.00	-2.33
	5	18	.44	-.21	.00	.00	.00	2.14
		19	-.44	.21	.00	.00	.00	-2.87
	6	18	.52	-.26	.00	.00	.00	2.35
		19	-.52	.26	.00	.00	.00	-3.25
	7	18	.59	-.31	.00	.00	.00	2.31
		19	-.59	.31	.00	.00	.00	-3.39
	8	18	.64	-.37	.00	.00	.00	1.91
		19	-.64	.37	.00	.00	.00	-3.17
	9	18	.66	-.42	.00	.00	.00	1.06
		19	-.66	.42	.00	.00	.00	-2.51

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
19	1	19	.04	-.02	.00	.00	.00	.28
		20	-.04	.02	.00	.00	.00	-.35
	2	19	.14	-.07	.00	.00	.00	1.00
		20	-.14	.07	.00	.00	.00	-1.25
	3	19	.24	-.12	.00	.00	.00	1.69
		20	-.24	.12	.00	.00	.00	-2.12
	4	19	.34	-.18	.00	.00	.00	2.33
		20	-.34	.18	.00	.00	.00	-2.93
	5	19	.43	-.23	.00	.00	.00	2.87
		20	-.43	.23	.00	.00	.00	-3.65
	6	19	.51	-.28	.00	.00	.00	3.25
		20	-.51	.28	.00	.00	.00	-4.21
	7	19	.58	-.34	.00	.00	.00	3.39
		20	-.58	.34	.00	.00	.00	-4.54
	8	19	.63	-.39	.00	.00	.00	3.17
		20	-.63	.39	.00	.00	.00	-4.52
	9	19	.65	-.45	.00	.00	.00	2.51
		20	-.65	.45	.00	.00	.00	-4.06
20	1	20	.04	-.02	.00	.00	.00	.35
		21	-.04	.02	.00	.00	.00	-.41
	2	20	.15	-.06	.00	.00	.00	1.25
		21	-.15	.06	.00	.00	.00	-1.47
	3	20	.25	-.11	.00	.00	.00	2.12
		21	-.25	.11	.00	.00	.00	-2.50
	4	20	.35	-.16	.00	.00	.00	2.93
		21	-.35	.16	.00	.00	.00	-3.47
	5	20	.44	-.21	.00	.00	.00	3.65
		21	-.44	.21	.00	.00	.00	-4.36
	6	20	.53	-.25	.00	.00	.00	4.21
		21	-.53	.25	.00	.00	.00	-5.09
	7	20	.60	-.31	.00	.00	.00	4.54
		21	-.60	.31	.00	.00	.00	-5.59
	8	20	.65	-.36	.00	.00	.00	4.52
		21	-.65	.36	.00	.00	.00	-5.75
	9	20	.67	-.42	.00	.00	.00	4.06
		21	-.67	.42	.00	.00	.00	-5.49
21	1	21	.04	-.02	.00	.00	.00	.41
		22	-.04	.02	.00	.00	.00	-.47
	2	21	.15	-.06	.00	.00	.00	1.47
		22	-.15	.06	.00	.00	.00	-1.68
	3	21	.25	-.10	.00	.00	.00	2.50
		22	-.25	.10	.00	.00	.00	-2.86
	4	21	.35	-.15	.00	.00	.00	3.47
		22	-.35	.15	.00	.00	.00	-3.99
	5	21	.45	-.19	.00	.00	.00	4.36
		22	-.45	.19	.00	.00	.00	-5.03
	6	21	.53	-.24	.00	.00	.00	5.09
		22	-.53	.24	.00	.00	.00	-5.92
	7	21	.60	-.29	.00	.00	.00	5.59
		22	-.60	.29	.00	.00	.00	-6.59
	8	21	.65	-.34	.00	.00	.00	5.75
		22	-.65	.34	.00	.00	.00	-6.94
	9	21	.68	-.40	.00	.00	.00	5.49
		22	-.68	.40	.00	.00	.00	-6.87

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
22	1	22	.04	-.02	.00	.00	.00	.47
		23	-.04	.02	.00	.00	.00	-.52
	2	22	.15	-.06	.00	.00	.00	1.68
		23	-.15	.06	.00	.00	.00	-1.88
	3	22	.25	-.10	.00	.00	.00	2.86
		23	-.25	.10	.00	.00	.00	-3.20
	4	22	.36	-.14	.00	.00	.00	3.99
		23	-.36	.14	.00	.00	.00	-4.47
	5	22	.45	-.18	.00	.00	.00	5.03
		23	-.45	.18	.00	.00	.00	-5.66
	6	22	.54	-.23	.00	.00	.00	5.92
		23	-.54	.23	.00	.00	.00	-6.71
	7	22	.61	-.27	.00	.00	.00	6.59
		23	-.61	.27	.00	.00	.00	-7.54
	8	22	.66	-.33	.00	.00	.00	6.94
		23	-.66	.33	.00	.00	.00	-8.07
	9	22	.69	-.38	.00	.00	.00	6.87
		23	-.69	.38	.00	.00	.00	-8.20
23	1	23	.04	-.01	.00	.00	.00	.52
		24	-.04	.01	.00	.00	.00	-.56
	2	23	.15	-.05	.00	.00	.00	1.88
		24	-.15	.05	.00	.00	.00	-2.00
	3	23	.26	-.09	.00	.00	.00	3.20
		24	-.26	.09	.00	.00	.00	-3.40
	4	23	.36	-.13	.00	.00	.00	4.47
		24	-.36	.13	.00	.00	.00	-4.77
	5	23	.46	-.16	.00	.00	.00	5.66
		24	-.46	.16	.00	.00	.00	-6.05
	6	23	.55	-.20	.00	.00	.00	6.71
		24	-.55	.20	.00	.00	.00	-7.19
	7	23	.62	-.25	.00	.00	.00	7.54
		24	-.62	.25	.00	.00	.00	-8.13
	8	23	.68	-.30	.00	.00	.00	8.07
		24	-.68	.30	.00	.00	.00	-8.77
	9	23	.70	-.35	.00	.00	.00	8.20
		24	-.70	.35	.00	.00	.00	-9.03
24	1	24	.02	-.04	.00	.00	.00	-.56
		25	-.02	.04	.00	.00	.00	.49
	2	24	.07	-.14	.00	.00	.00	-2.00
		25	-.07	.14	.00	.00	.00	1.75
	3	24	.13	-.24	.00	.00	.00	-3.40
		25	-.13	.24	.00	.00	.00	2.98
	4	24	.18	-.34	.00	.00	.00	-4.77
		25	-.18	.34	.00	.00	.00	4.18
	5	24	.23	-.43	.00	.00	.00	-6.05
		25	-.23	.43	.00	.00	.00	5.30
	6	24	.29	-.51	.00	.00	.00	-7.19
		25	-.29	.51	.00	.00	.00	6.31
	7	24	.34	-.58	.00	.00	.00	-8.13
		25	-.34	.58	.00	.00	.00	7.13
	8	24	.40	-.62	.00	.00	.00	-8.77
		25	-.40	.62	.00	.00	.00	7.69
	9	24	.46	-.64	.00	.00	.00	-9.03
		25	-.46	.64	.00	.00	.00	7.91

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
25	1	25	.02	-.04	.00	.00	.00	-.49
		26	-.02	.04	.00	.00	.00	.37
	2	25	.07	-.14	.00	.00	.00	-1.75
		26	-.07	.14	.00	.00	.00	1.32
	3	25	.13	-.24	.00	.00	.00	-2.98
		26	-.13	.24	.00	.00	.00	2.26
	4	25	.18	-.34	.00	.00	.00	-4.18
		26	-.18	.34	.00	.00	.00	3.16
	5	25	.23	-.43	.00	.00	.00	-5.30
		26	-.23	.43	.00	.00	.00	4.01
	6	25	.29	-.51	.00	.00	.00	-6.31
		26	-.29	.51	.00	.00	.00	4.77
	7	25	.34	-.58	.00	.00	.00	-7.13
		26	-.34	.58	.00	.00	.00	5.40
	8	25	.40	-.62	.00	.00	.00	-7.69
		26	-.40	.62	.00	.00	.00	5.82
	9	25	.46	-.64	.00	.00	.00	-7.91
		26	-.46	.64	.00	.00	.00	5.99
26	1	26	.02	-.04	.00	.00	.00	-.37
		27	-.02	.04	.00	.00	.00	.25
	2	26	.07	-.14	.00	.00	.00	-1.32
		27	-.07	.14	.00	.00	.00	.90
	3	26	.13	-.24	.00	.00	.00	-2.26
		27	-.13	.24	.00	.00	.00	1.53
	4	26	.18	-.34	.00	.00	.00	-3.16
		27	-.18	.34	.00	.00	.00	2.15
	5	26	.23	-.43	.00	.00	.00	-4.01
		27	-.23	.43	.00	.00	.00	2.72
	6	26	.29	-.51	.00	.00	.00	-4.77
		27	-.29	.51	.00	.00	.00	3.24
	7	26	.34	-.58	.00	.00	.00	-5.40
		27	-.34	.58	.00	.00	.00	3.66
	8	26	.40	-.62	.00	.00	.00	-5.82
		27	-.40	.62	.00	.00	.00	3.95
	9	26	.46	-.64	.00	.00	.00	-5.99
		27	-.46	.64	.00	.00	.00	4.06
27	1	27	.02	-.04	.00	.00	.00	-.25
		28	-.02	.04	.00	.00	.00	.13
	2	27	.07	-.14	.00	.00	.00	-.90
		28	-.07	.14	.00	.00	.00	.47
	3	27	.13	-.24	.00	.00	.00	-1.53
		28	-.13	.24	.00	.00	.00	.81
	4	27	.18	-.34	.00	.00	.00	-2.15
		28	-.18	.34	.00	.00	.00	1.13
	5	27	.23	-.43	.00	.00	.00	-2.72
		28	-.23	.43	.00	.00	.00	1.43
	6	27	.29	-.51	.00	.00	.00	-3.24
		28	-.29	.51	.00	.00	.00	1.70
	7	27	.34	-.58	.00	.00	.00	-3.66
		28	-.34	.58	.00	.00	.00	1.93
	8	27	.40	-.62	.00	.00	.00	-3.95
		28	-.40	.62	.00	.00	.00	2.08
	9	27	.46	-.64	.00	.00	.00	-4.06
		28	-.46	.64	.00	.00	.00	2.14

61

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
28	1	28	.02	-.04	.00	.00	.00	-.13
		29	-.02	.04	.00	.00	.00	.00
2	28	28	.07	-.14	.00	.00	.00	-.47
		29	-.07	.14	.00	.00	.00	.00
3	28	28	.13	-.24	.00	.00	.00	-.81
		29	-.13	.24	.00	.00	.00	.00
4	28	28	.18	-.34	.00	.00	.00	-1.13
		29	-.18	.34	.00	.00	.00	.00
5	28	28	.23	-.43	.00	.00	.00	-1.43
		29	-.23	.43	.00	.00	.00	.00
6	28	28	.29	-.51	.00	.00	.00	-1.70
		29	-.29	.51	.00	.00	.00	.00
7	28	28	.34	-.58	.00	.00	.00	-1.93
		29	-.34	.58	.00	.00	.00	.00
8	28	28	.40	-.62	.00	.00	.00	-2.08
		29	-.40	.62	.00	.00	.00	.00
9	28	28	.46	-.64	.00	.00	.00	-2.14
		29	-.46	.64	.00	.00	.00	.00

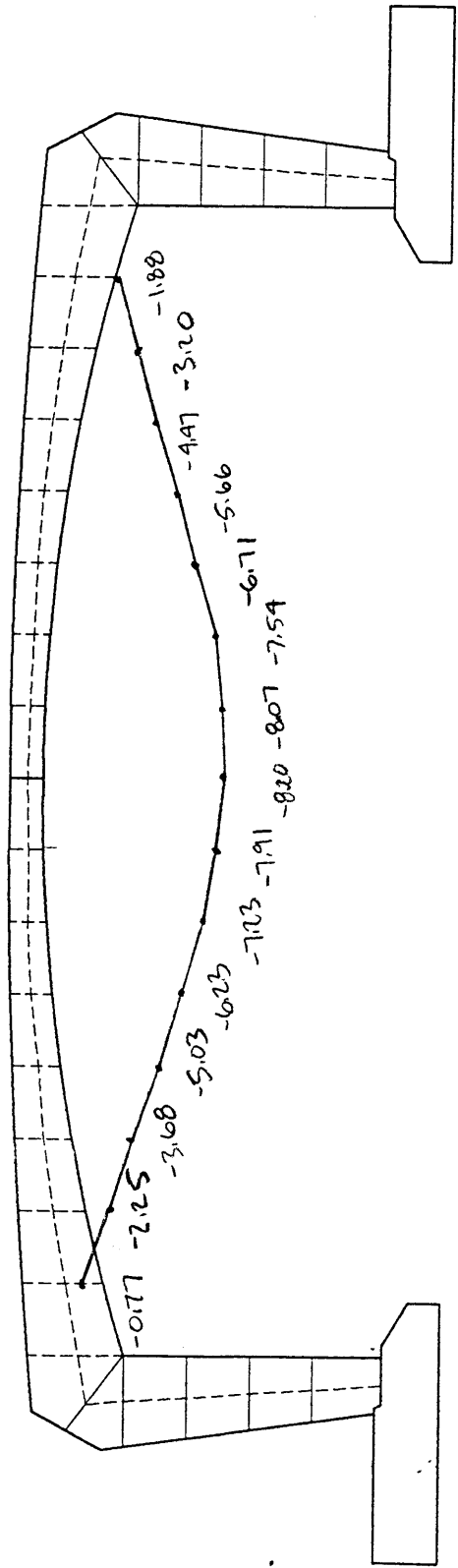
\*\*\*\*\* END OF LATEST ANALYSIS RESULT \*\*\*\*\*

\*\*\* STAAD-III ERROR MESSAGE \*\*\*  
 NO END OR FINISH STATEMENT FOUND. RUN TERMINATED.

\*\*\*\*\* END OF STAAD-III \*\*\*\*\*

\*\*\*\* DATE= AUG 28,1998 TIME= 8:23:31 \*\*\*\*

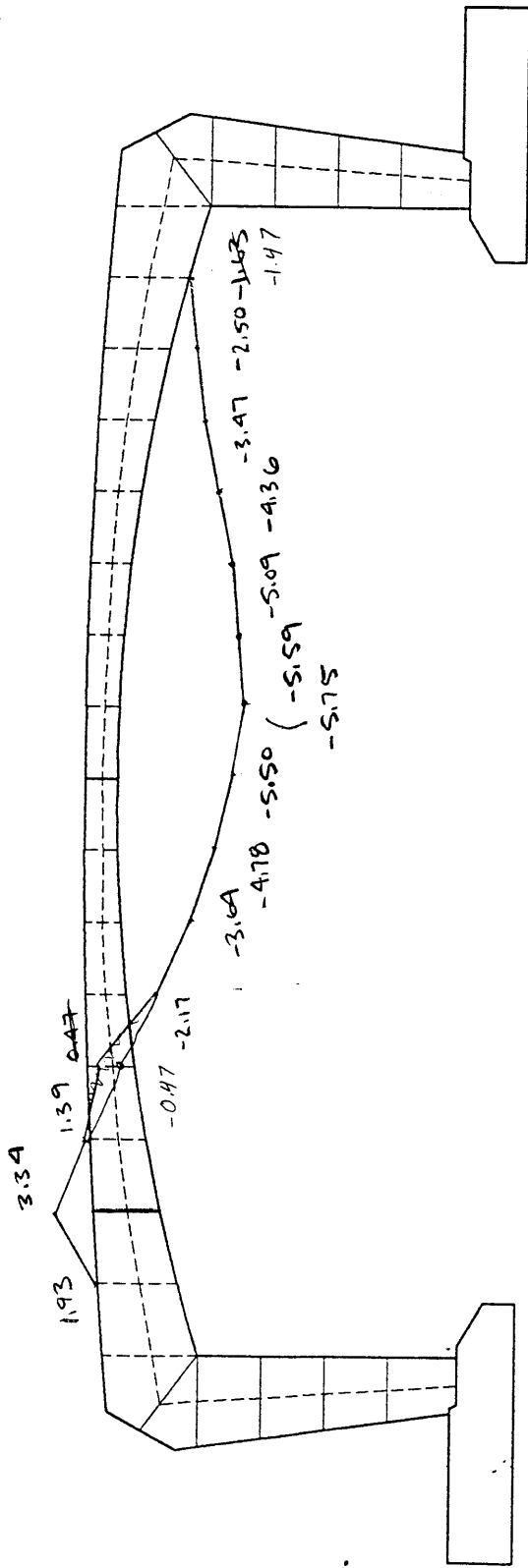
\*\*\*\*\*  
 \* For questions on STAAD-III, contact: \*  
 \* Research Engineers, Inc at \*  
 \* West Coast: Ph- (714) 974-2500 Fax- (714) 921-2543 \*  
 \* East Coast: Ph- (508) 688-3626 Fax- (508) 685-7230 \*  
 \*\*\*\*\*



CEMETERY ROAD UNDERPASS

INFLUENCE LINE FOR JOINT 7

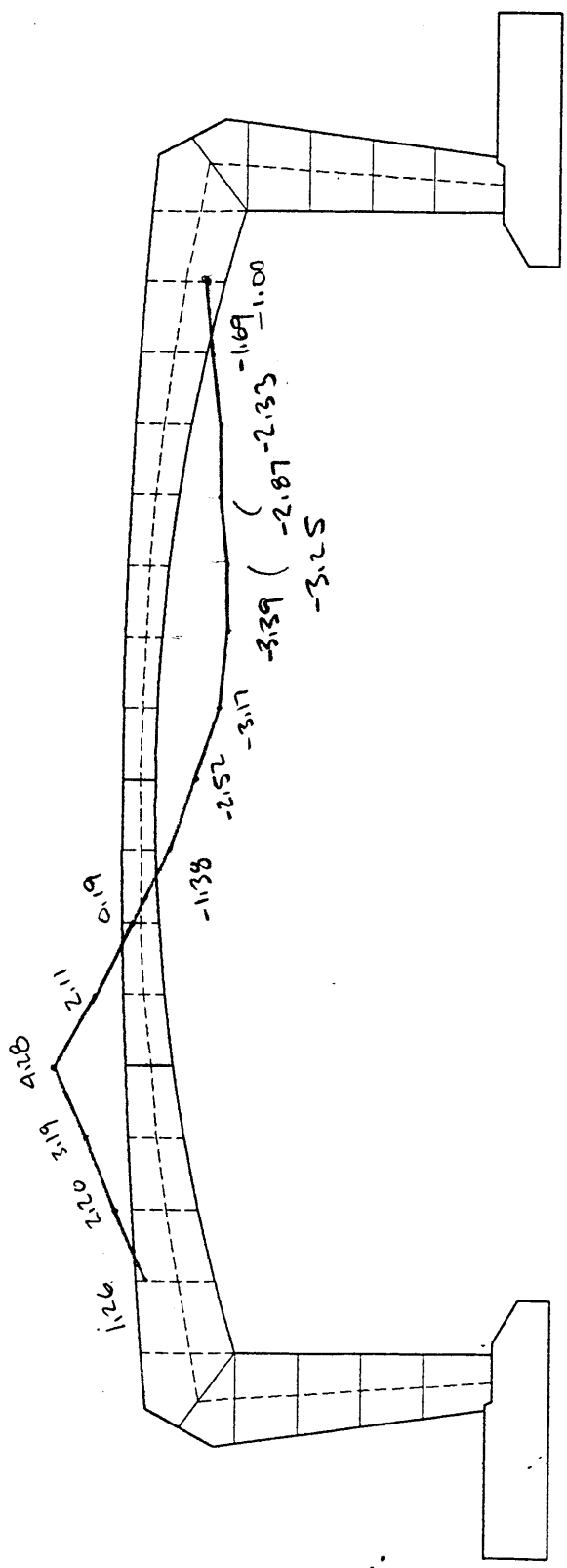
(FACE OF ABUTMENT)



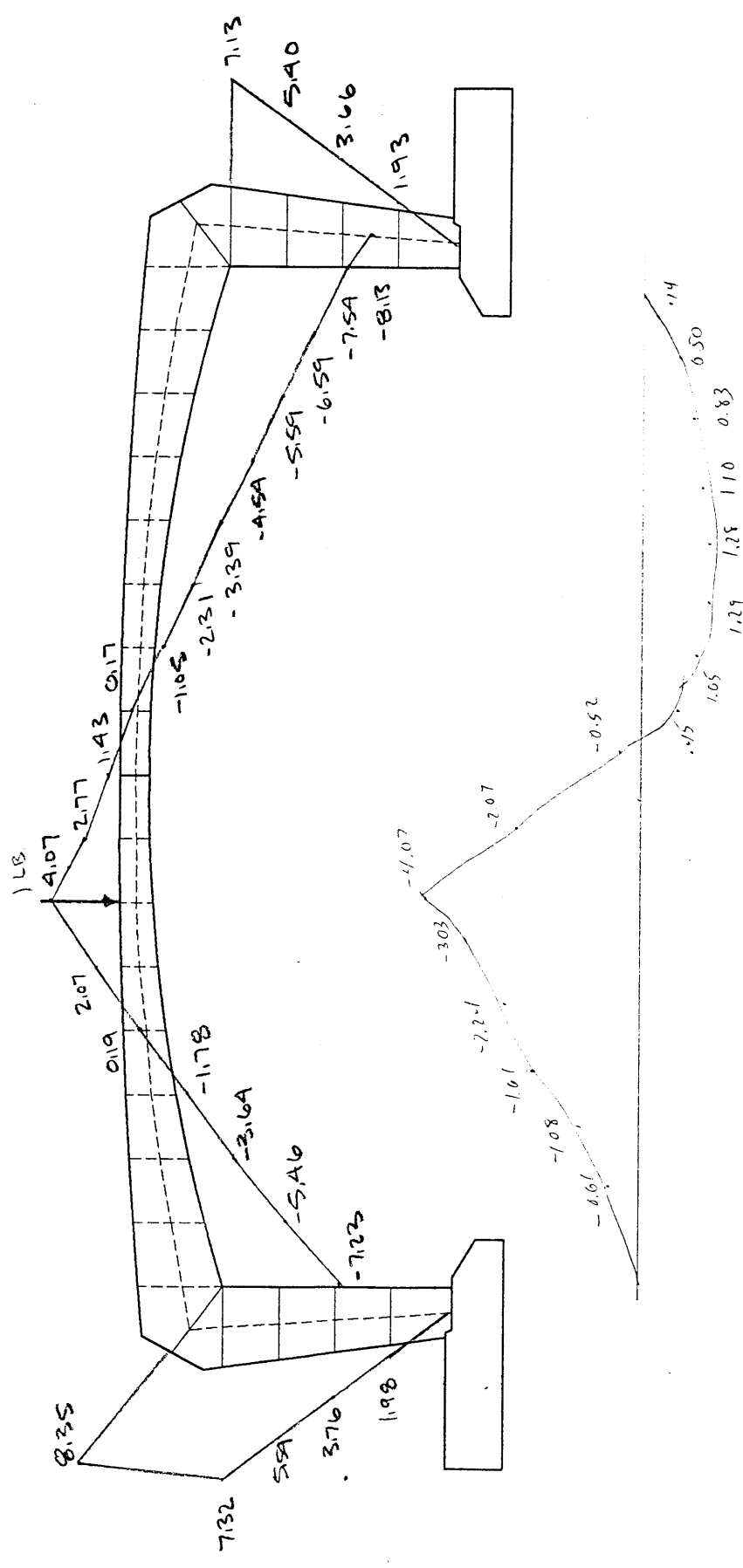
CEMETERY ROAD UNDERPASS

INFLUENCE LINE FOR JOINT 9 (1/8th POINT)



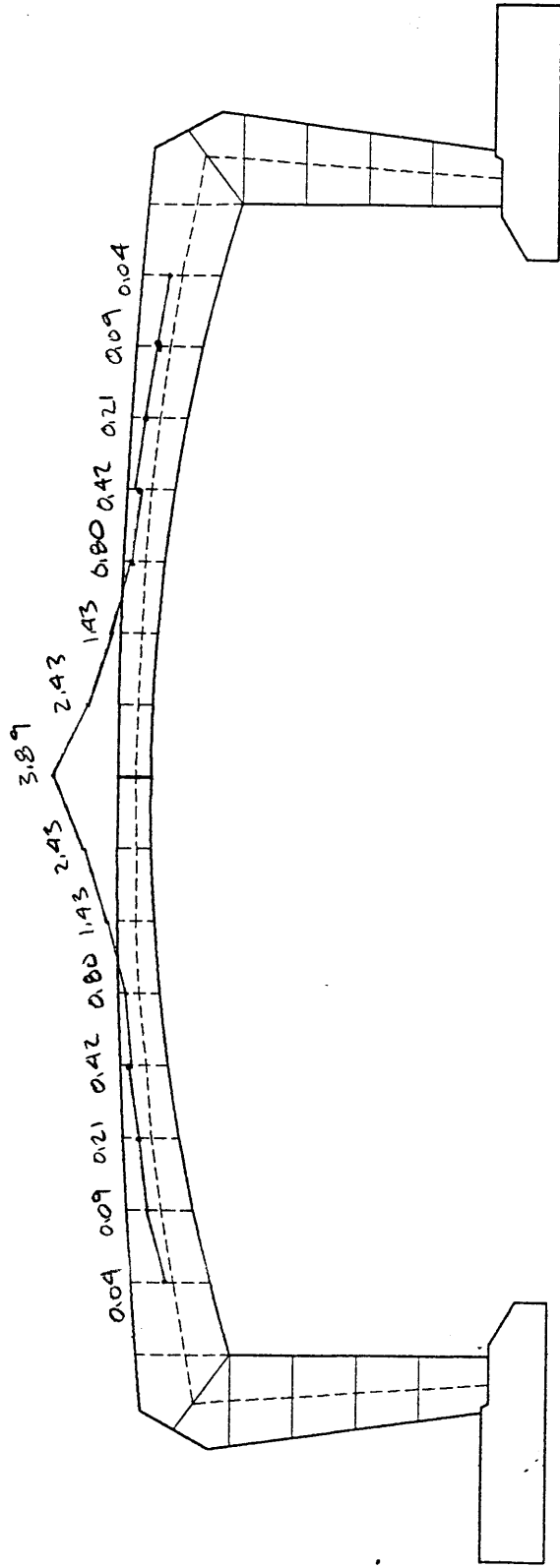


CEMETERY ROAD UNDERPASS  
INFLUENCE LINE FOR JOINT 11 (1/4 POINT)



# CEMETERY ROAD UNDERPASS

INVERSE LINE OVER JOINT 13 (3/8th Point)



CEMETERY ROAD UNDERPASS

INFLUENCE LINE FOR JOINT 15 (ROWN)